

Fast and Efficient
Design for Stability

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walter
p moore

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Session topics

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Stability-design overview – AISC Methods
Story Drift and Drift Limits – *wind and seismic*
Indirect Analysis Method – **NEW!**
Background of the AISC R_M factor
Second-order drift methods
Example

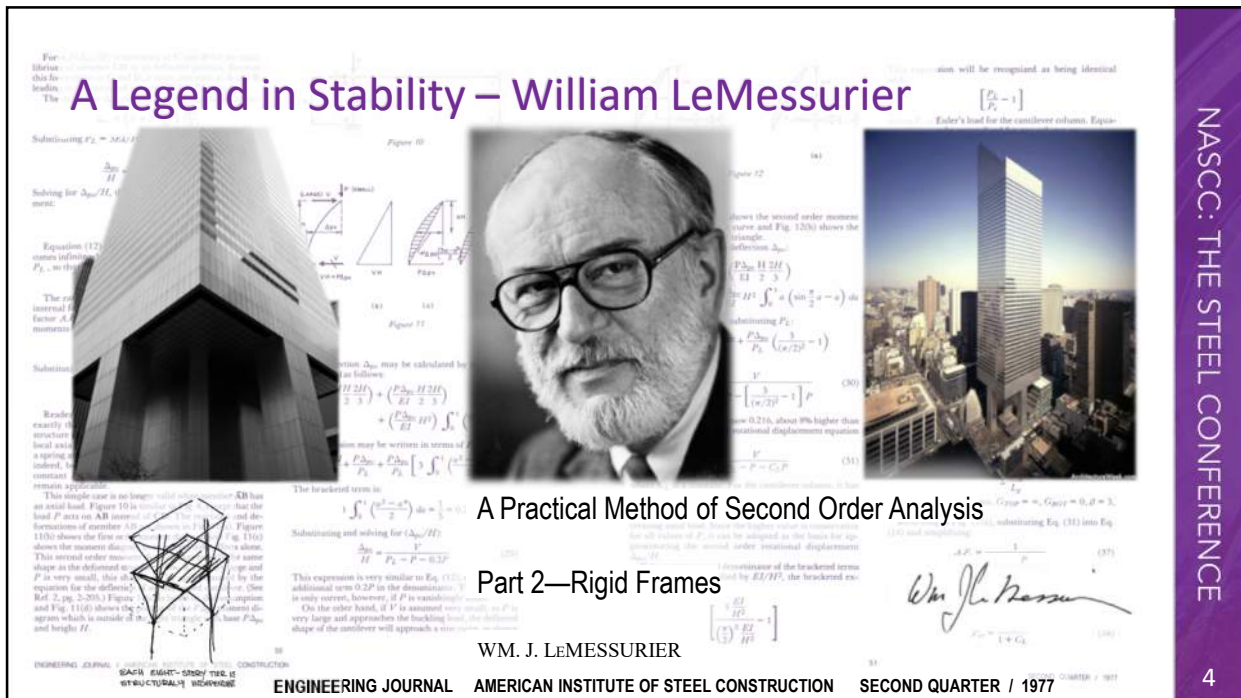
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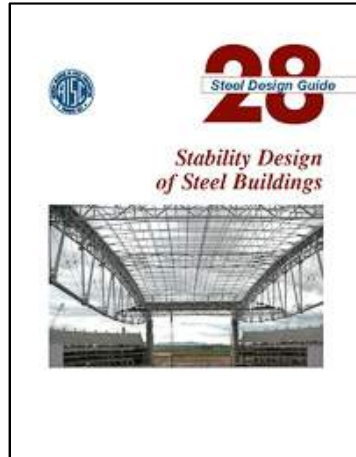


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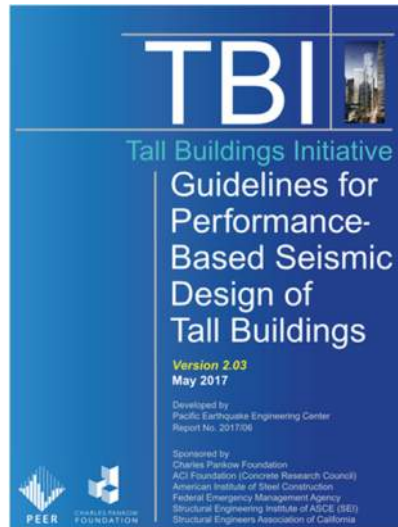
4

AISC DESIGN GUIDE 28



Authors:
Don White
Larry Griffis

Performance Based Seismic Design Tall Buildings - PEER



Stability and Instability – SSRC Guide

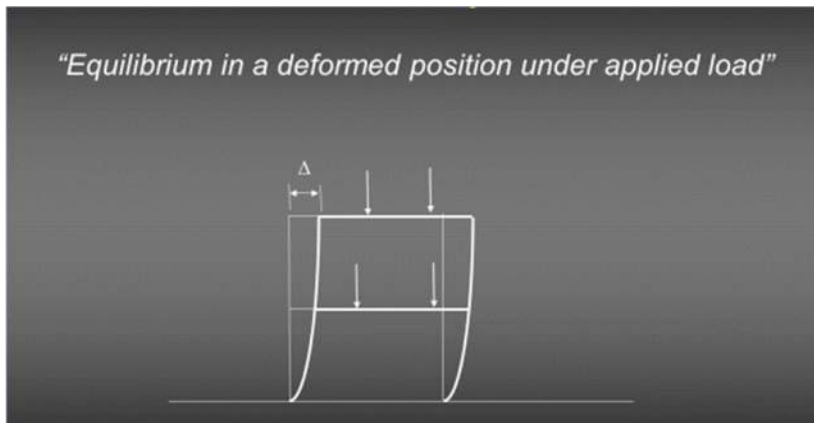
Stability
(SSRC Guide – Galambos 1998)

- *Stability* – the capacity of a compression member, element or frame to remain in position and support load, even if forced slightly out of line or position by an added lateral force
- *Instability* – a condition reached during buckling under increasing load in a compression member, element or frame at which the capacity for resistance to added load is exhausted and continued deformation results in a decrease in load carrying capacity

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What is stability?

“Equilibrium in a deformed position under applied load”



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Design for stability

Design considerations

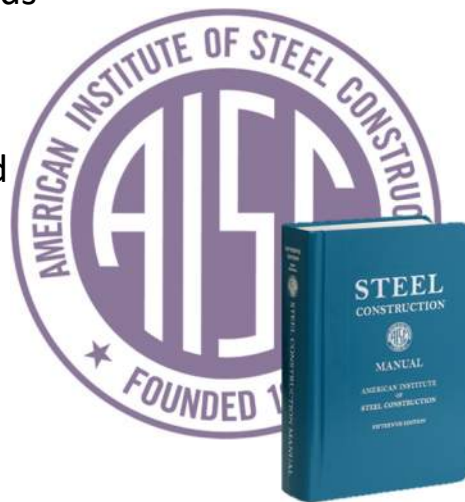
- Strength
 - Adequate to resist loads
- Stiffness
 - Prevent excessive displacement
- Stability
 - Not a separate consideration
 - Increases strength demand
 - Decreases stiffness

Design for stability

General requirements for all methods

AISC-defined methods

- FOM: First-order method
- ELM: Effective length method
- DM: Direct analysis method



Stability Requirements “The Big 5” (Chapter C)

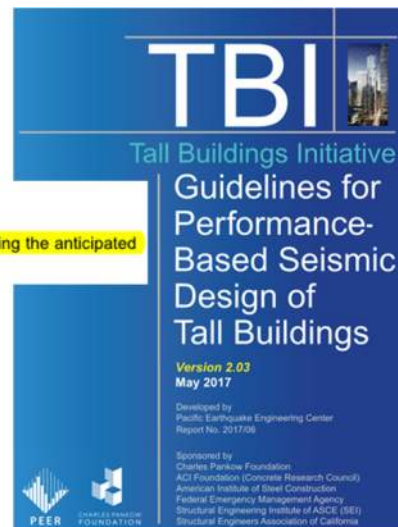
- *All deformations considered*
 - *flexural, shear, axial, panel zone, etc.*
- *Second order analysis*
 - *P-Δ, P-δ effects*
- *Geometric imperfections*
 - *Out-of-straightness*
 - *Out-of-plumbness*
- *Member stiffness reduction due to residual stresses*
 - *EI reduction due to premature yielding*
- *Uncertainty in stiffness strength*

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Performance Based Design The Analysis Model

4.6 STRUCTURAL MODELING PARAMETERS

Develop the structural analysis model based on expected properties considering the anticipated level of response and damage.



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Consider all deformations

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Axial
Flexural
Shear

Figure 3-12. Panel zone inelastic deformation (exaggerated).

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Geometric imperfections

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$L/500$
Out-of-plumbness

$L/1000$
Out-of-straightness

Connection flexibility

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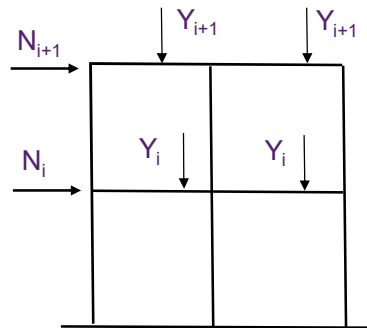
14

What are “notional loads”?

“Lateral loads applied at each story expressed as a fraction (0.002) of the gravity load at a story”

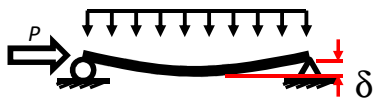
$$N_i = \sum 0.002 Y_i$$

Now part of DM and ELM!



$P\delta$ (member) vs $P\Delta$ (system)

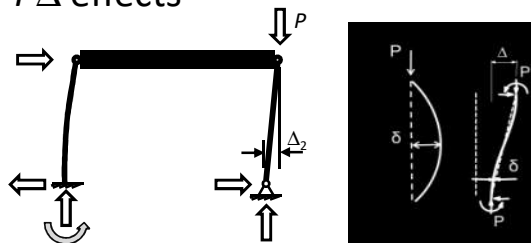
$P\delta$ effects



$P\Delta$ = Additional moment due to the member axial force acting thru the relative transverse displacement of the member ends

$P\delta$ = additional moment due to the member axial force acting thru the transverse displacement of the cross-section relative to a chord between the member ends

$P\Delta$ effects



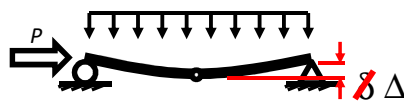
Note: δ influences Δ

$P\delta$ modeling

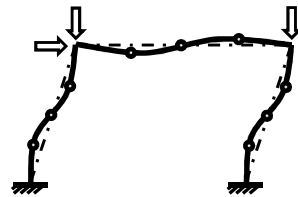
Software typically addresses $P\Delta$ effects at nodes

$P\delta$ effects can be modeled

- Introduce nodes: $\delta \rightarrow \Delta$

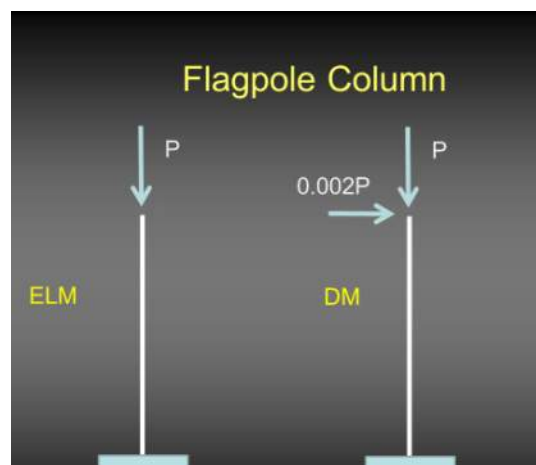


- 4 segments/member



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Notional load for the Direct Analysis Method



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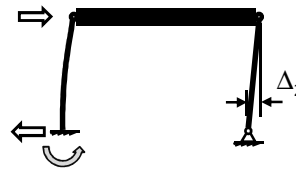
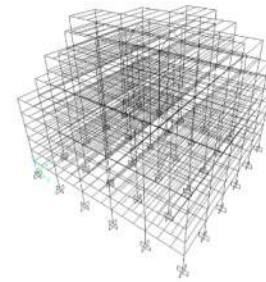
Big 5 considerations

All deformations

- Typically addressed in modeling

Second-order effects

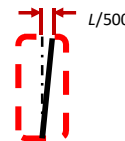
- Second-order analysis



5 considerations

Geometric imperfections

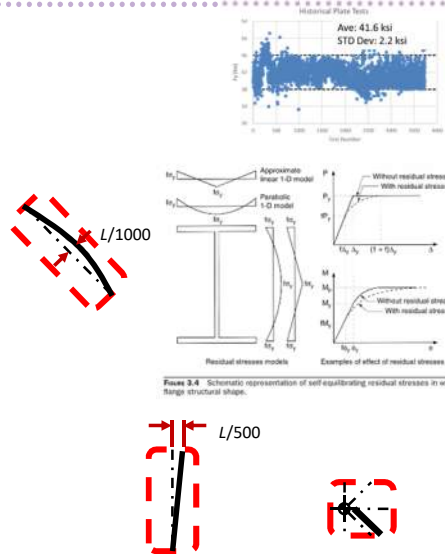
- Member
 - Reduced design strength
 - Modeling of imperfections
- System
 - Notional load
 - Modeling of imperfections



5 considerations

Stiffness reduction due to inelasticity & uncertainty in strength and stiffness

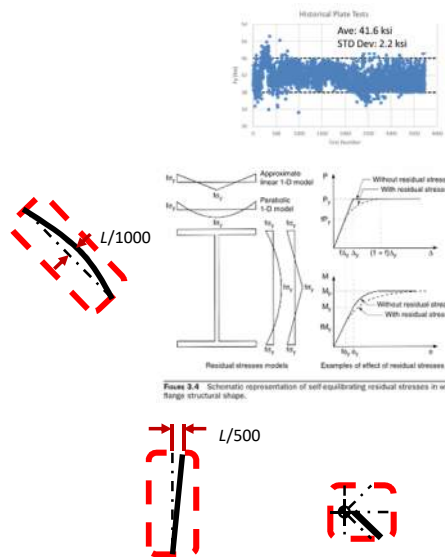
- Member
 - Reduced strength
 - ϕ
 - Modeling



5 considerations

Stiffness reduction due to inelasticity & uncertainty in strength and stiffness


- System
 - Reduced stiffness
 - Additional load
 - Amplified load
 - Modeling



Accounting for out-of-straightness

Geometric Imperfections
How are they accounted for ?

Out-of-Straightness of members (L/1000)



L

- Accounted for in ELM and DM in Column Equations

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Accounting for out-of-straightness

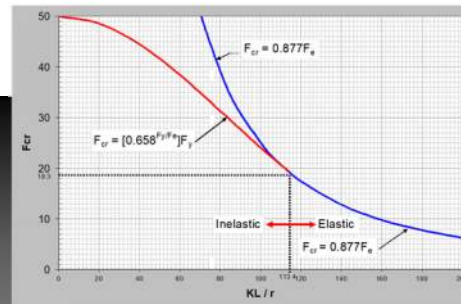
AISC Column Equations

$$P_n = F_{cr} A_g$$

$$F_e = \pi^2 E / (KL/r)^2$$

Reduced from 1.0 for out-of-straightness

$$F_{cr} = 0.877 F_e \text{ elastic}$$

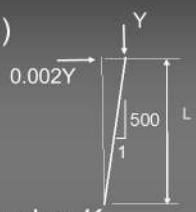
$$F_{cr} = [0.658^{F_y/F_e}] F_y \text{ inelastic}$$


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Accounting for out-of-plumbness

Geometric Imperfections
How are they accounted for ?

Out-of-Plumbness ($L/500$)



- Accounted for in ELM by using K
- Accounted for in DM by using *notional loads* or, modeling out-of-plumb structure directly


Stiffness reduction due to inelasticity

Residual stress

$$M_y < M_p$$

Residual Stresses Influence Stability

$f_r = 0.3F_y$ compression at flange tips



1. $E_T = tE$ Tangent Modulus or
2. $I = tI$ stiffness of elastic core

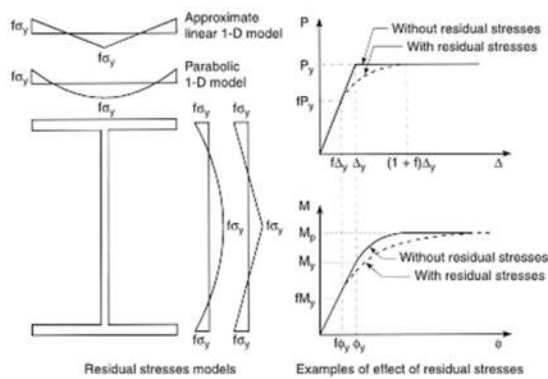


FIGURE 3.4 Schematic representation of self-equilibrating residual stresses in wide-flange structural shape.

First-order (analysis) method: FOM

First-order analysis

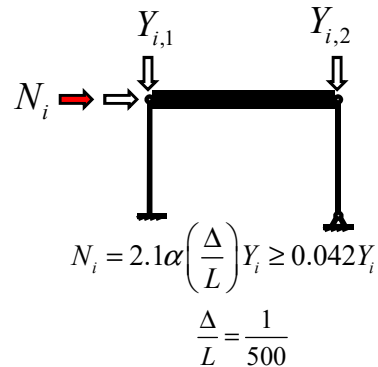
Limit on second-order effects

- $B_2 \leq 1.5$

Additional lateral load

- Based on
 - $B_2 = 1.5$
 - Geometric imperfection
 - Reduction in stiffness due to inelasticity
 - Uncertainty in strength and stiffness

Use for simple structures
Use for quick design check
Conservative solutions



The ELM

- Analyze ideal geometrically perfect elastic structure
- Account for residual stresses & geometric imperfections *implicitly with K (buckling analysis)*
- Calculate $K > 1$ for MF's (or obtain elastic column buckling load P_e from a sidesway buckling analysis) - $K = 1.0$ for braced frames
- Use the AISC column curve to determine P_n (can be expressed in terms of elastic buckling stress $F_e = P_e / A_g$)

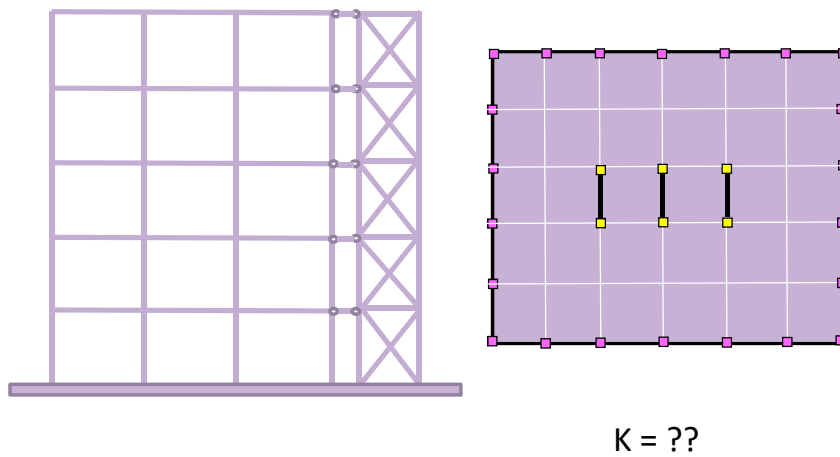
The K Factor

Finding K can be complicated!

rigid link
pin ends
leaning columns
Find K?

$$K_{\text{story}} = \sqrt{\frac{1}{P_e} \frac{\sum P_e + \sum C_i P_e}{\sum P_e}}$$

Moment Frame–Braced Frames combined



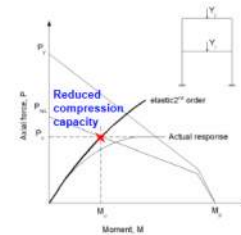
Direct analysis method: DM

Second-order analysis

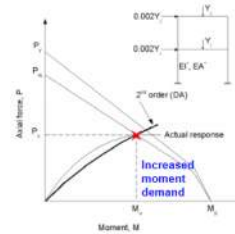
Reduced-stiffness model

K = 1!

- Stiffness reduction due to inelasticity
- Uncertainty in strength and stiffness



Effective Length Method



Direct Analysis Method

Direct analysis method: DM

Stiffness reduction

- General
 - $0.8EI$
 - $0.8EA$
- Flexural columns $\alpha P_r / P_{ns} \geq 0.5$ (50% of yield)
 - $0.8\tau_b EI$

K = 1!

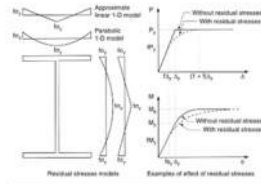
$$\tau_b = 4 \frac{\alpha P_r}{P_{ns}} \left(1 - \frac{\alpha P_r}{P_{ns}} \right) \leq 1.0$$

- Not applicable to
 - *Building period*
 - *Drift*

Direct analysis method: DM

Stiffness reduction

- General
 - $0.8EI$
 - $0.8EA$
- Flexural columns $\alpha P_r / P_{ns} \geq 0.5$ (50% of yield)
 - $0.8\tau_b EI$
- Not applicable to
 - *Building period*
 - *Drift*



K = 1!

Figure 3.4 Schematic representation of self-equilibrating residual stresses in wide-flange structural shapes.

$$\tau_b = 4 \frac{\alpha P_r}{P_{ns}} \left(1 - \frac{\alpha P_r}{P_{ns}} \right) \leq 1.0$$

ELM vs DM

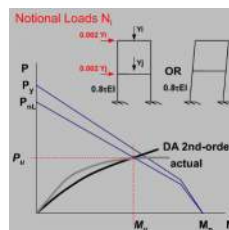
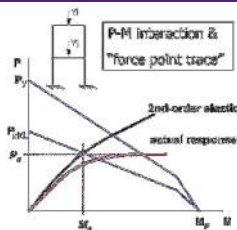
Interaction Beam Column Equation

$$\text{for } \frac{F_r}{F_c} \geq 0.2 \quad \frac{F_r}{F_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0 \quad \text{H1-1a}$$

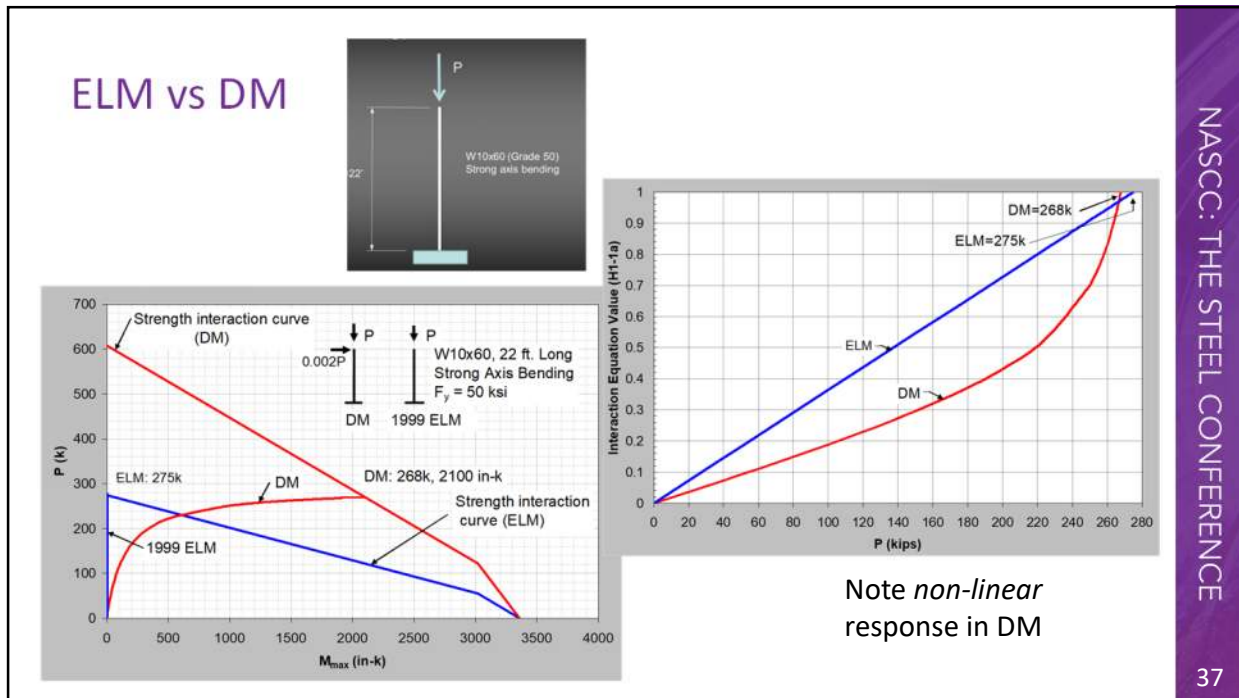
ELM Use of $K > 1$ causes smaller P_e , larger P_r / P_c Axial dominates

DM $K=1$ causes larger P_e , smaller P_r / P_c
notional load makes larger M / M_c Moment dominates

Effective Length Method



Direct Analysis Method



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- ### Fundamental differences: ELM vs DM
- Geometric imperfections
 - (out-of-plumbness) included *explicitly* in the DM analysis using *notional loads* or *modeling out-of-plumb geometry*
 - *Reduced stiffness* of structure used in DM analysis (accounts for softening of the structure at ultimate load from residual stresses)

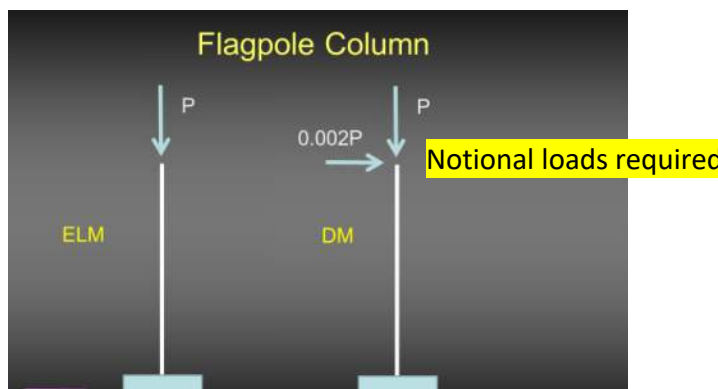
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Major advantages of the Direct Analysis Method

- **No K factors** are required!
- Internal forces are more accurate
- Applies to all frame types – moment frames, braced frames, combined systems
- More economical beam-column proportions in certain cases
- The DM now the preferred method in Chapter C

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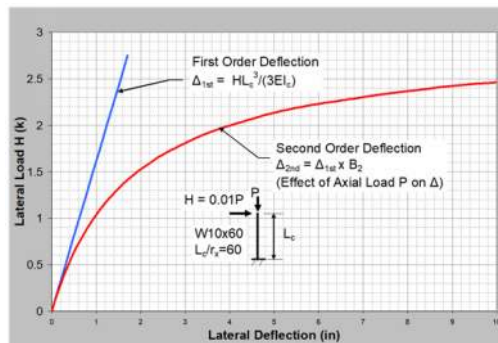
ELM vs DM – Modeling difference



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Second-Order Analysis is important

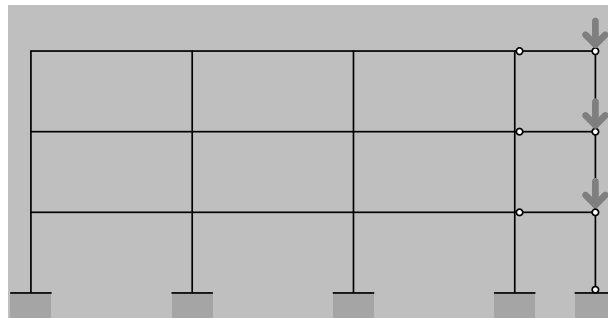
- Requires input of gravity loads into analysis model
- *Without gravity loads in model, 2nd order analysis is the same as 1st order analysis!*



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Include **ALL** gravity & other loads that influence the structure's stability

- ... the loads in any gravity columns, walls, etc. that are stabilized by the lateral load resisting system must be included in the analysis
- This is often handled by the use of a "dummy column"



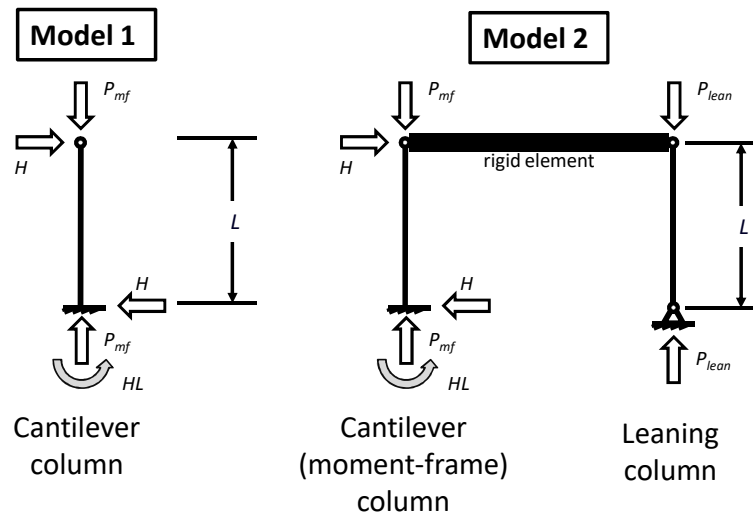
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Notional loads also apply to the ELM now

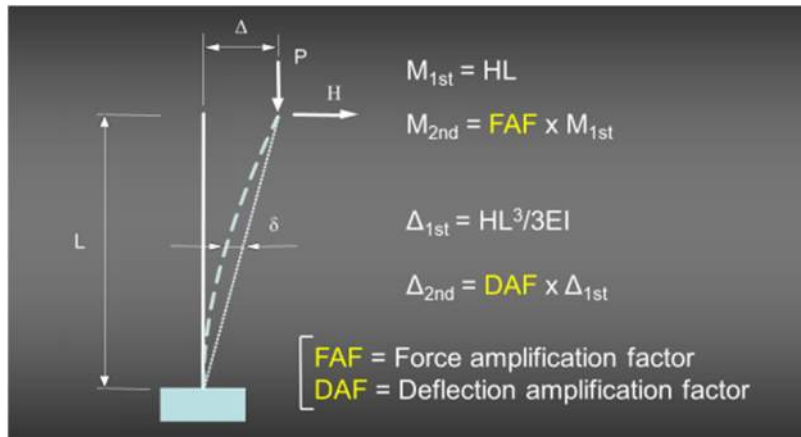
New restrictions on the ELM:

- Sidesway amplification limited to 1.5
 - $(\Delta_{2nd} / \Delta_{1st} \leq 1.5$ based on the nominal unreduced stiffness or
 - $\Delta_{2nd} / \Delta_{1st} \leq 1.71$ based on the reduced stiffness)
- A notional minimum lateral load is required in gravity-only load combinations (to account for the effect of nominal out-of-plumbness)

Simple Stability Models – Useful Tools



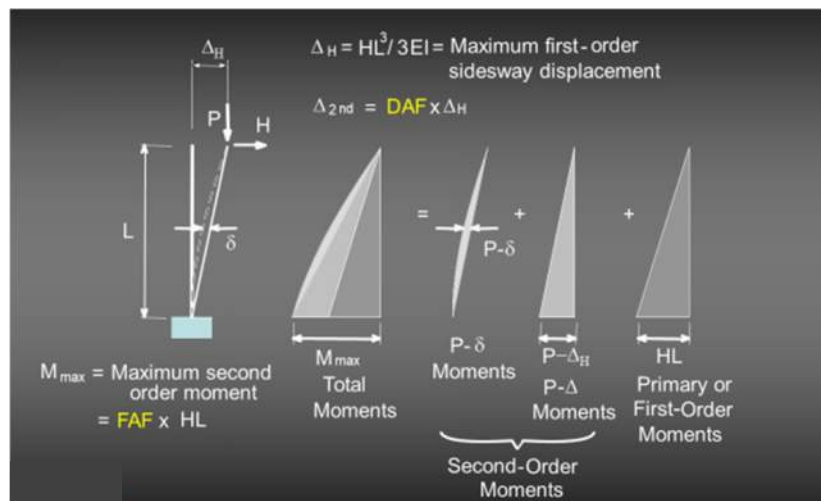
Cantilever column – Model 1



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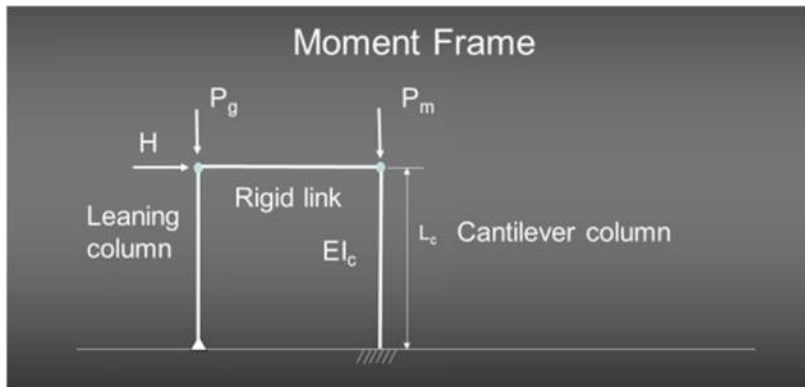
Cantilever column – Model 1



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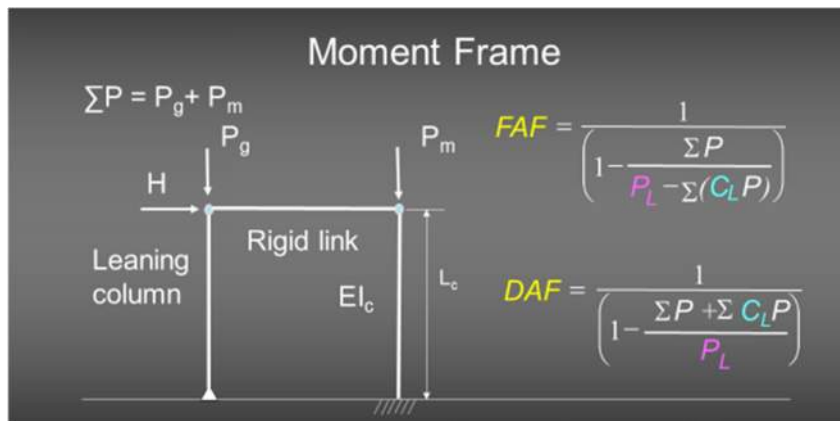
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Basic Stability Model – Model 2 (LeMessurier 1977)



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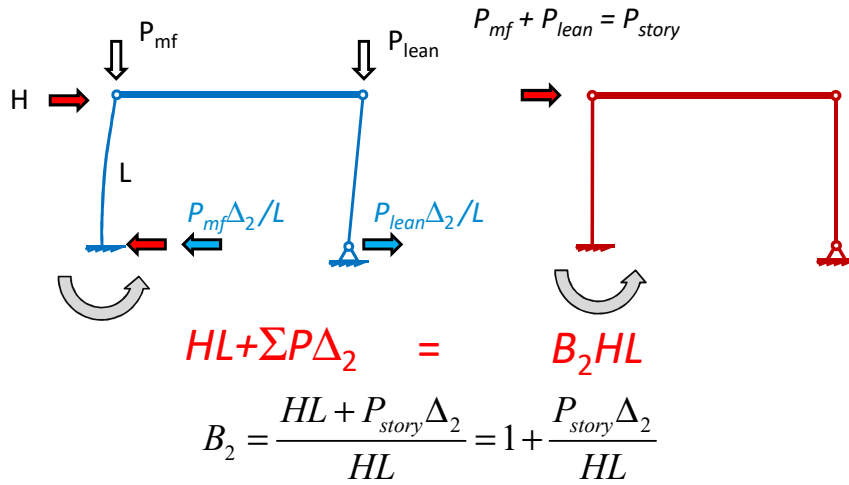
Stability Model 2 (LeMessurier – 1977)



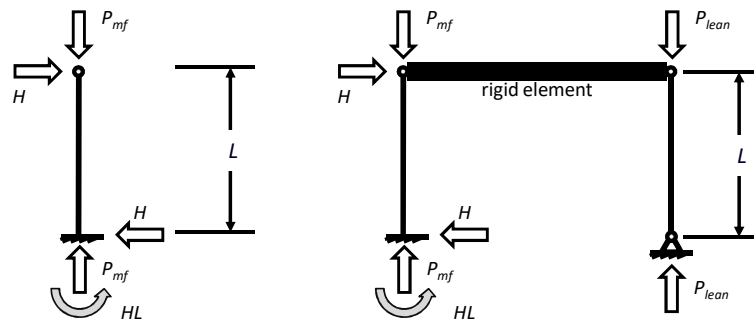
FAF ≈ DAF but not exactly the same!

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Second-order analysis

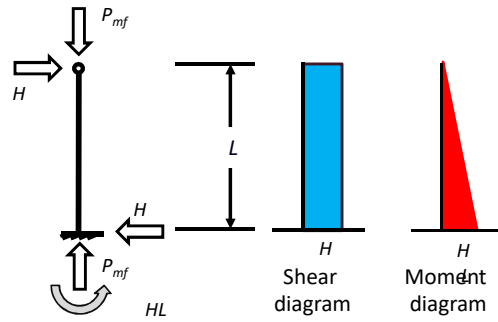


First-order analysis



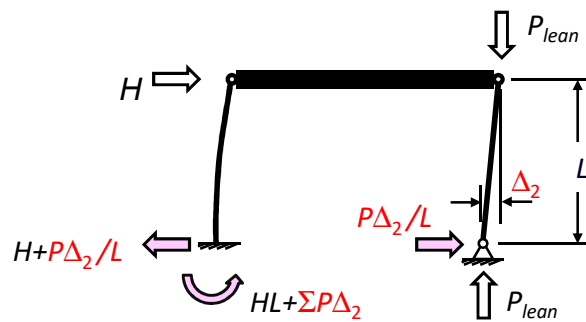
Inclusion of gravity load does not affect first-order analysis

First-order analysis



Inclusion of gravity load does not affect first-order analysis

Second-order analysis



Leaning-column effect
Equilibrium in the deformed condition (Δ_2)

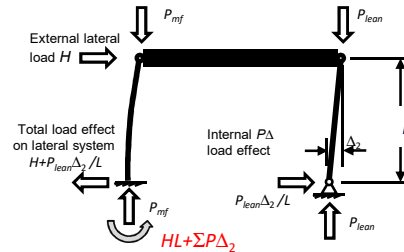
Second-order analysis

Lateral-load-effect amplifier:

$$HL + \sum P\Delta_2 = B_2 HL$$

Gravity load increases lateral-load effects

$$B_2 = \frac{HL + \sum P\Delta_2}{HL} = 1 + \frac{\sum P\Delta_2}{HL}$$

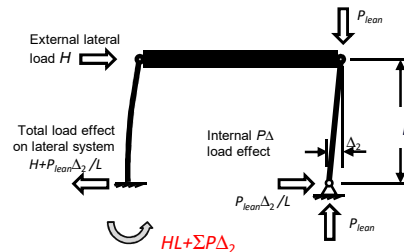


Second-order analysis

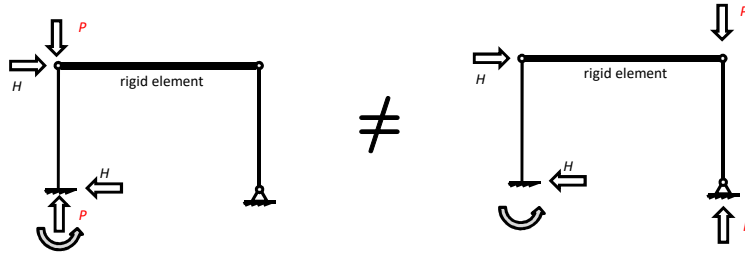
$$B_2 = 1 + \frac{\sum P\Delta_2}{HL}$$

$$\Delta_2 = B_2 \Delta_1 \quad (\text{true if there is no } P\delta \text{ effect})$$

$$B_2 = \frac{1}{1 - \frac{\sum P\Delta_1}{HL}}$$



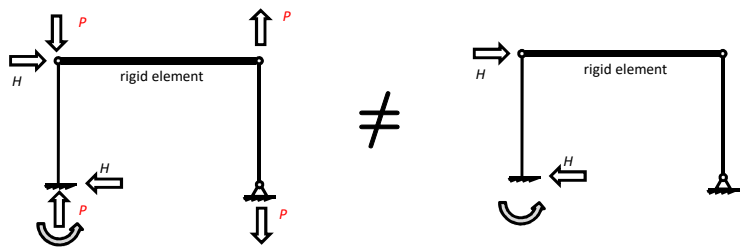
Pδ stiffness reduction



$P\delta + P\Delta$ stiffness reduction > $P\Delta$ stiffness reduction

Important for moment-frame structures

Pδ stiffness reduction



$(P\delta + P\Delta) - P\Delta$ stiffness reduction > zero

Important for moment-frame structures

Pδ stiffness reduction

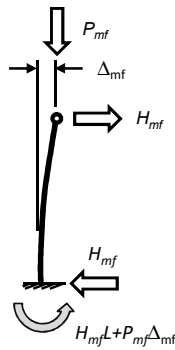
$$B_2 = \frac{1}{1 - \frac{P_{story}}{HL/\Delta_1 - (12/\pi^2 - 1)P_{mf}}} \leq \frac{1}{1 - \frac{P_{story}\Delta_1}{\left(1 - 0.15 \frac{P_{mf}}{P_{story}}\right)HL}}$$

LeMessurier, 1977 AISC Specification

Note: AISC parameter R_M is a conservative value appropriate for force amplification

Gravity load on flexural columns has greater effect than gravity load on leaning columns

- Moment-frame columns
- Cantilever columns

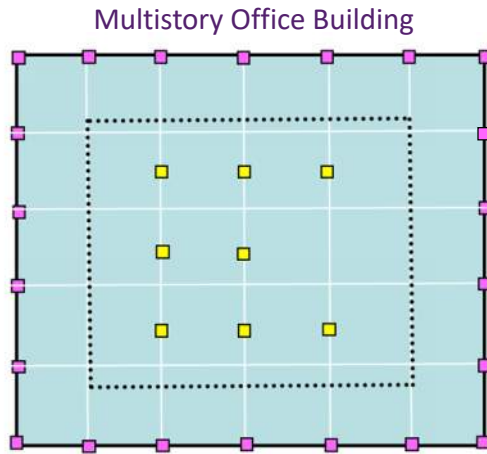


Importance of the “leaning columns”



*47 Story Office Building
Houston, Texas*

Effect of “leaning columns”



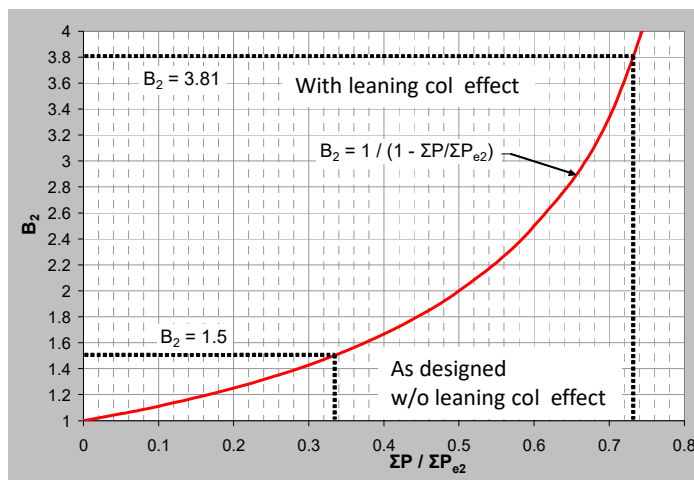
ΣP = all gravity load

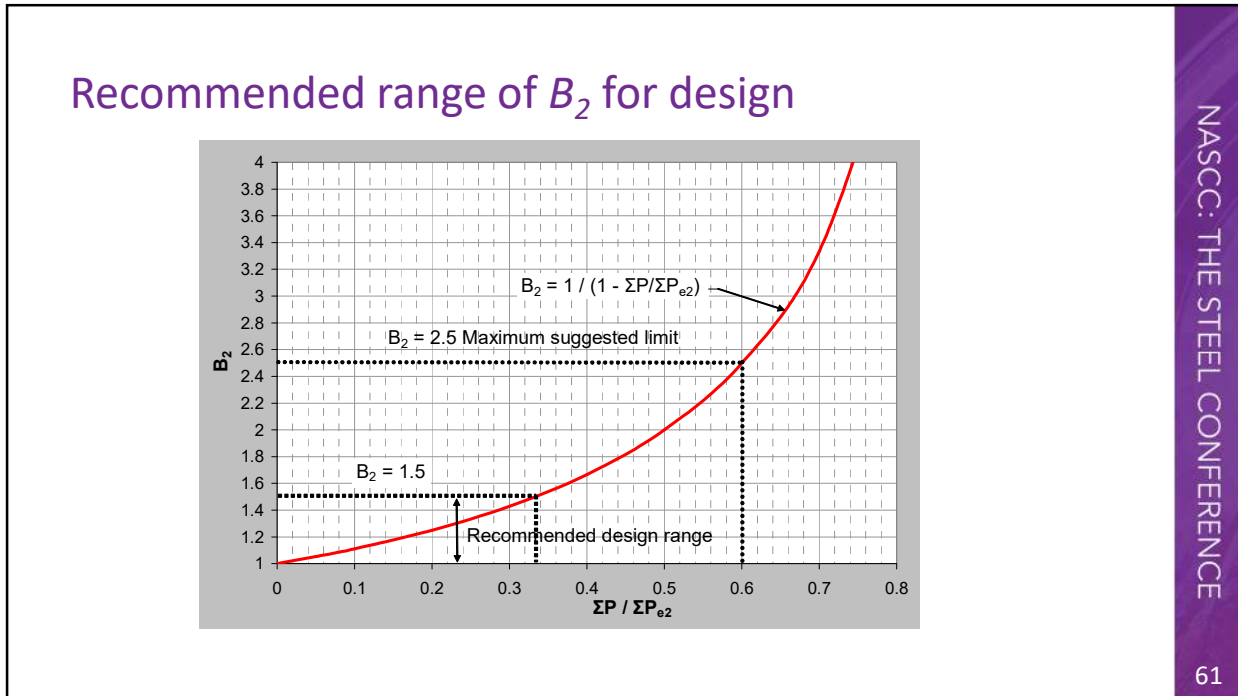
$$P_{ext} = 0.45 \Sigma P$$

$$P_{int} = 0.55 \Sigma P$$

MF - Lateral Load Resisting System around building perimeter (30' bays)

Effect of “leaning columns”





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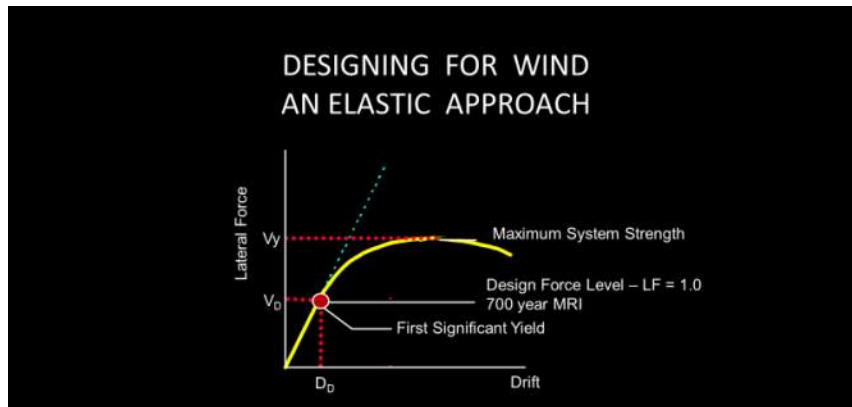
Limit States - including Drift Control

- Limit State 1: Strength under *ultimate wind*
- Limit State 2: Serviceability under *service level wind*
- Limit State 3: Strength under *design seismic*
- Limit State 4: Stability (Drift) under *design seismic*
- Limit State 5: Serviceability under *service level seismic*

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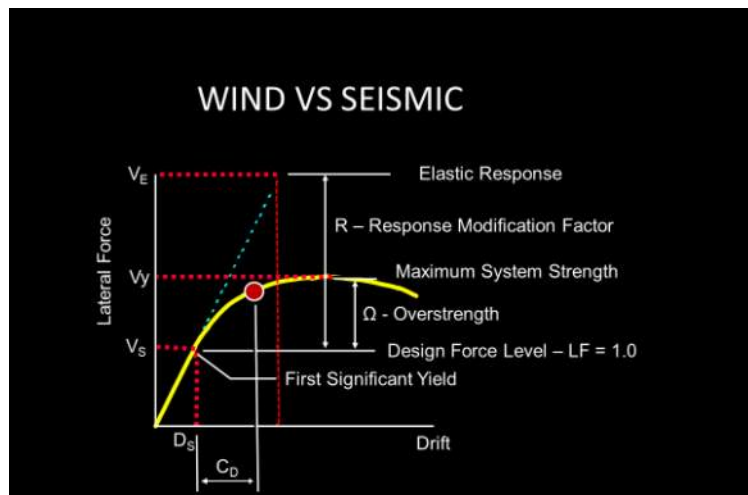
Designing for wind load



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Wind vs Seismic



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Design Limit States

**LIMIT STATES
TALL BUILDING DESIGN**

- **Strength Limit State**
 - Prescribed by Building Codes – life-safety
 - Strong enough to resist gravity, wind, seismic loads
- **Serviceability Limit State**
 - Story Drift (Deflection under lateral load)
 - Perception to motion under lateral load “sway” (measured by acceleration in milli-g’s)
 - Engineering judgment (not a code requirement)

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Some key factors for building design

THREE KEY FACTORS

- **Strength** under wind and seismic loads
- **Stiffness** to control interstory drift (deflection)
- **Dynamic Properties** (mass, stiffness, damping) to control perception to wind motion

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Designing for Wind

CURRENT APPROACH DESIGNING FOR WIND

Limit States (Performance Levels):

- Strength: MRI = 700, 1700 Years (Code)
- Story Drift: MRI = 10, 25, 50, 100 Years
(Engr. Judgment)
- Perception to Motion: MRI = 1, 10 Years
(Engr. Judgment)

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The importance of the wind tunnel

If you want to understand how
real buildings behave
under wind storms.....

Go to the Wind Tunnel!

frequencies,
mode shapes,
mass,
stiffness,
damping
"True Building Response"



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Damage control under wind load



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Serviceability Limit States can control design

SERVICEABILITY LIMIT STATES

- **IMPORTANT NOTE:**
- *SERVICEABILITY LIMIT STATES (e.g. deflections, vibration) often control the design of many building floor members – particularly in lighter steel framed buildings.*
- *Many designers will design for serviceability limit states first – and then check strength limit states as a final step in the design.*

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Requirements for assessing building drift & damage control

- Build an accurate analysis model
- Define one or more serviceability load combinations
- Define a damage measure that captures the damage potential from building sway

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The change in member properties

Serviceability limit state to *Strength* limit state load levels

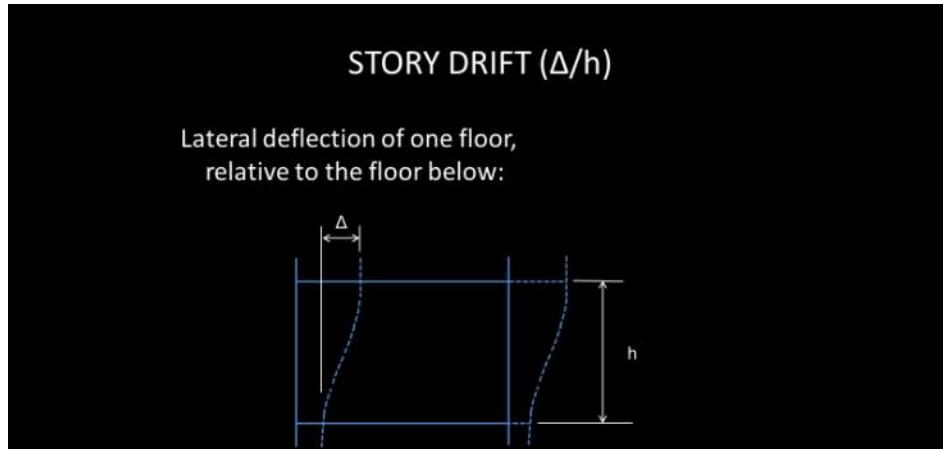
- GROSS MEMBER PROPERTIES – under serviceability load combinations, structure remains essentially elastic
- REDUCED MEMBER PROPERTIES, reduced stiffness, under strength limit states, structures in inelastic range of response
- Applies to steel structures, more so to concrete structures
- Applies to building Period T
- Applicable in wind tunnel studies
- **Must consider this effect in design**

**TWO SEPARATE MODELS
– ONE FOR EACH LIMIT STATE**

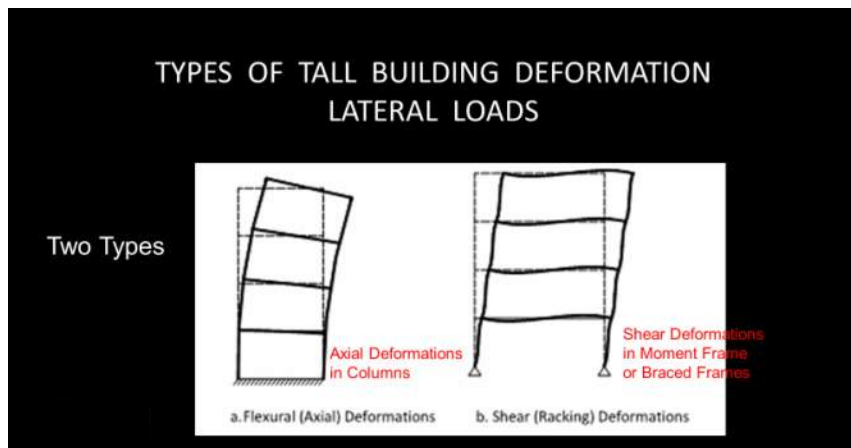
A B_3 Factor can change this!

72

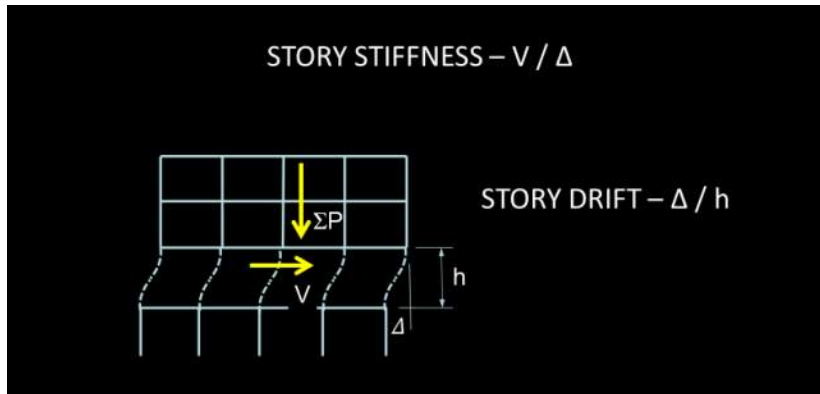
Story Drift – a popular measure of deflection control



Deformation components

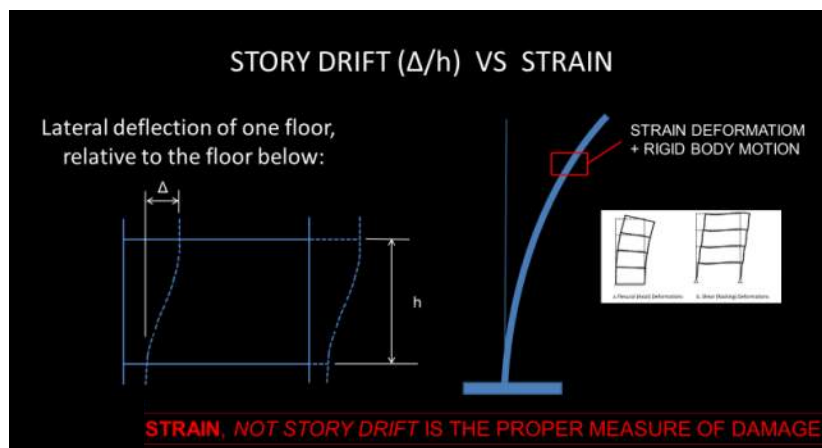


“Racking” deformation

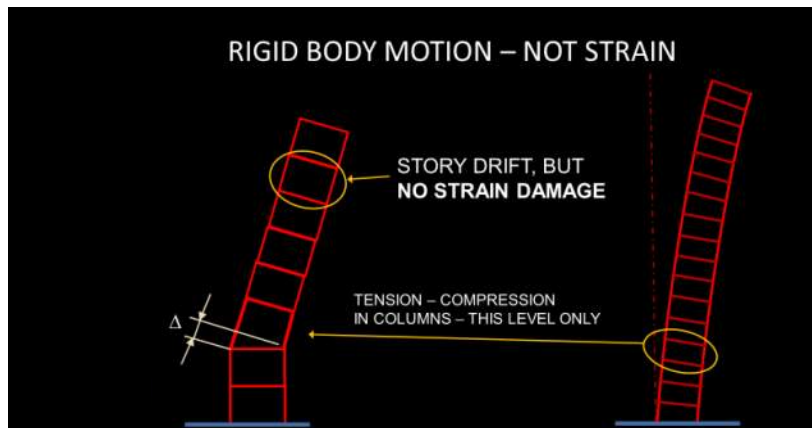


Important for moment frame systems with wide column spacing

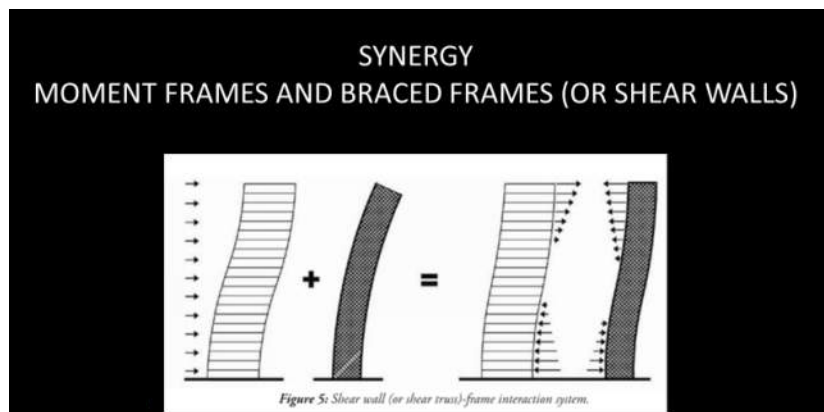
Story Drift vs Strain



Rigid body motion *does not* cause strain



Moment Frame – Braced Frame interaction



Designing for *Drift* and *Perception to Motion*

**THE NEW YORK SLENDER TOWER STORY
CONDOS -BILLIONAIRE'S ROW**

111 W 57th ST
84 stories,
1428 ft
h/w = 23/1



435 Park Av.
85 stories,
1396 ft
h/w = 15/1



Strength to Serviceability Ratio

STRENGTH VS SERVICEABILITY FORCES/DEFLECTIONS EXAMPLES

- Kansas City – Typical inland city (middle of range US wind city)
 $V_{700} = 110$ MPH; $V_{25} = 79$ MPH; V^2 Ratio = 1.94
- Los Angeles – West coast city (lower bound US wind city)
 $V_{700} = 115$ MPH; $V_{25} = 84$ MPH; V^2 Ratio = 1.87
- Miami – High Hurricane city (upper bound range US wind city)
 $V_{700} = 170$ MPH; $V_{25} = 113$ MPH; V^2 Ratio = 2.26

STR / SERV (700/25)	LOS ANGELES	KANSAS CITY	MIAMI
FORCE - CONCRETE	2.24	2.33	2.71
LATERAL Δ - CONCRETE	3.19	3.31	3.85
FORCE - STEEL	2.24	2.33	2.71
LATERAL Δ - STEEL	2.80	2.91	3.39

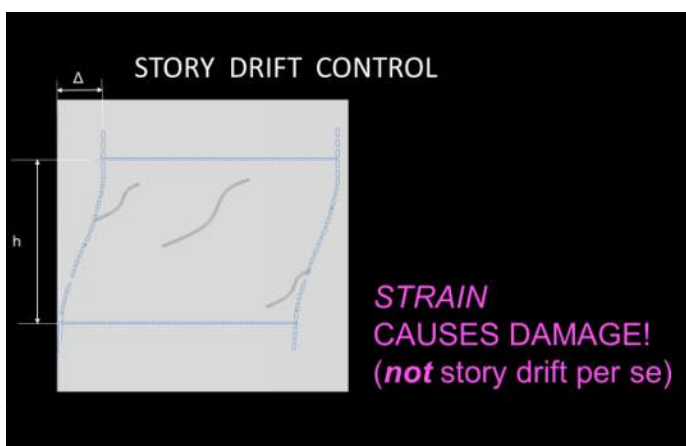
Story Drift

STORY DRIFT (Δ/h)

- Choose demand wind storm
 - 10, 25, 50, 100 yr MRI (ASCE 7 App C)
- Choose Drift Limit (material, cladding type)
 - $h/100$ – $h/800$ limits common ($1/400 = 0.0025$)
- Accurate computer model
 - Ultimate, service level
 - Proper stiffness selection (cracking)
 - Consider all deformations applicable (axial, flexural, shear, panel zone, P- Δ effects)

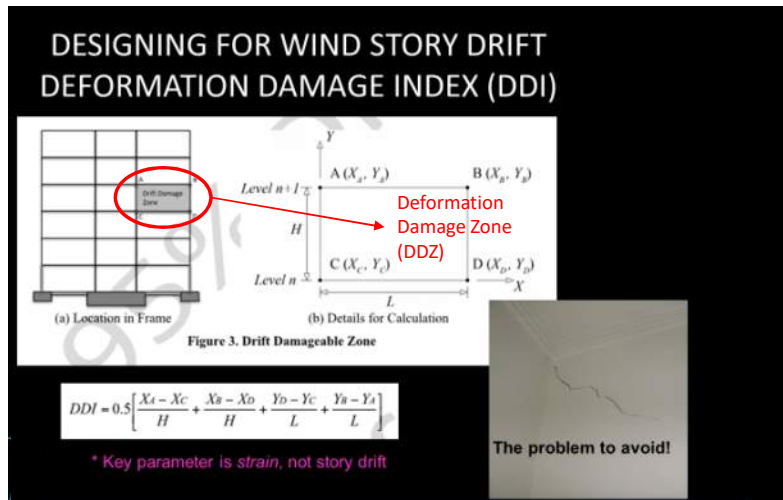
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Story Drift vs Strain



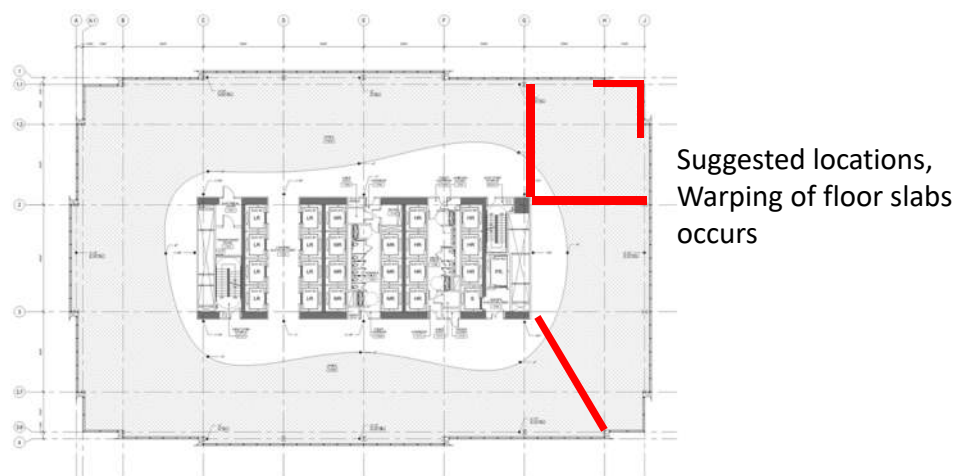
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It's really a matter of *strain* – *not drift*



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Deformation Damage Zones (DDZ)



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Suggested Drift Design Criteria – ATC Design Guide

SUGGESTED DRIFT DESIGN CRITERIA DEFORMATION DAMAGE INDEX (DDI)

Table 2
Suggested Minimum Building Serviceability (Drift) Criteria under Wind Load

Building Quality and Durability	Serviceability Wind Load	DDI (Strain)	Comment
Minimum	10 yr. MRI	0.0025	Building defined by Stakeholders desire for a <i>minimum</i> standard level of quality and durability at least cost.
Normal	25 yr. MRI	0.0025	Building defined by Stakeholders desire for a <i>typical or mid-level</i> standard of quality and durability.
High	50 yr. MRI	0.002	Building defined by Stakeholders desire for a <i>high level</i> standard of quality and durability.
Premium	100 yr. MRI	0.002	Building defined by Stakeholders desire for a <i>premium level</i> standard of quality and durability.

Irwin, Peter
Griffis, Larry

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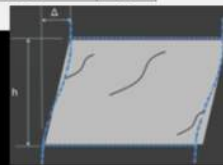
The Deformation Damage Index Limits

DEFORMATION DAMAGE INDICES (DDI)

Table 3. Suggested DDI Limits for Serviceability Design

Building Element	Suggested DDI Limit	Notes	
Exterior Cladding	Brick Veneer w/ metal studs	0.0025	1
	Brick Veneer w/ unreinf. masonry	0.0025	1,2
	Plaster/Stucco	0.0025	3
	Architectural Precast	0.0025	4
	Stone Clad Precast	0.0025	4
	Architectural Metal Panel	0.01	5
Curtain Wall, Window Wall		0.0025	6
		0.0025	7
Interior Partitions	Gypsum Drywall, Plaster	0.00150	8
	Concrete Masonry, unreinf.	0.00150	8
	Tile, Hollow Clay Brick	0.0005	9
Elevators	Drywall enclosure	0.0025	10

The problem to avoid!



FRAGILITY CURVE (FEMA P-58)

Fragility Curve – a mathematical relationship between an engineering demand parameter (e.g. shear strain in a nonstructural component) and the probability of attaining damage

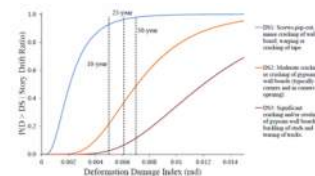


Figure 7. Gypsum Wall Board Fragility Curves (PACT)

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AISC 360/341 vs. ASCE-7 symbols

$$B_2 = \frac{1}{1 - \theta / R_M}$$

$$B_2 = \frac{1}{1 - \frac{\alpha P_{story} \Delta_H}{R_M H L}}$$

$K_M = \frac{H}{\Delta_H} = \frac{V_x I_e}{\Delta / C_d}$

Mechanical stiffness Δ / C_d

$$\theta = \frac{P_x \Delta I_e}{V_x h_{sx} C_d}$$

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Seismic design for drift and stability – ASCE 7-16

- ASCE 7-16 Section 12.12 Drift and Deformation
- Table 12.12-1 Allowable Story Drift Δ_{all}

$$\theta = \frac{P_x \Delta I_e}{V_x h_{sx} C_d} \quad (12.8-16) \quad \text{STABILITY INDEX } \theta$$

$$\theta_{max} = \frac{0.5}{\beta C_d} \leq 0.25 \quad (12.8-17)$$

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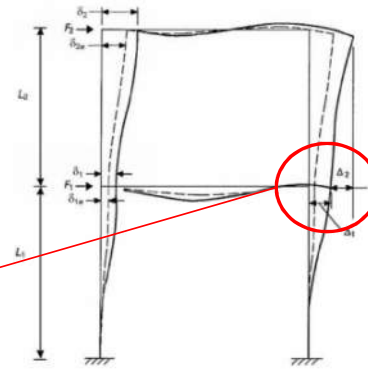
Seismic design for drift and stability – ASCE 7-16

- Design Story Drift Δ :

$$\delta_x = \frac{C_d \delta_{xe}}{I_e} \quad (12.8-15)$$

$\delta_2 = C_d \delta_{2e} / I_E =$ amplified displacement; (Table 12.12-1).

$$\Delta_2 = C_d (\delta_{2e} - \delta_{1e}) / I_E \leq \Delta_d$$



Drift Control under ultimate design loads

Seismic Drift Limits – ASCE 7 Table 12.12-1

Table 12.12-1 Allowable Story Drift, $\Delta_d^{a,b}$

Structure	Risk Category		
	I or II	III	IV
Structures, other than masonry shear wall structures, four stories or less above the base as defined in Section 11.2, with interior walls, partitions, ceilings, and exterior wall systems that have been designed to accommodate the story drifts	$0.025h_{sx}^c$	$0.020h_{sx}$	$0.015h_{sx}$
Masonry cantilever shear wall structures ^d	$0.010h_{sx}$	$0.010h_{sx}$	$0.010h_{sx}$
Other masonry shear wall structures	$0.007h_{sx}$	$0.007h_{sx}$	$0.007h_{sx}$
All other structures	$0.020h_{sx}$	$0.015h_{sx}$	$0.010h_{sx}$

Performance Based Design The Analysis Model

4.6 STRUCTURAL MODELING PARAMETERS

Develop the structural analysis model based on expected properties considering the anticipated level of response and damage.

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Performance Based Design – Seismic Tall Buildings

Deformation-controlled action – An action expected to undergo nonlinear behavior in response to earthquake shaking, and which is evaluated for its ability to sustain such behavior.

Force-controlled action – An action that is not expected to undergo nonlinear behavior in response to earthquake shaking, and which is evaluated on the basis of its available strength.

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Structural Analysis Model

4.6 STRUCTURAL MODELING PARAMETERS

Develop the structural analysis model based on expected properties considering the anticipated level of response and damage.

Commentary: The structural analysis model is intended to provide an unbiased, best estimate of expected response of the building when subjected to earthquake ground motion. For this reason, the structural analysis model should be developed based on expected material and member stiffnesses, strengths, and deformation capacities, rather than values that have been adjusted to achieve a “conservative” estimate of response.

Service Level Earthquake – PEER (MRI = 43 years)

Service-Level Earthquake (SLE) shaking.

LINEAR DYNAMIC ANALYSIS

- Response Spectrum Analysis
- Response History Analysis

Define SLE shaking as required in Chapter 3, with a minimum return period of 43 years (50% probability of exceedance in 30 years). Represent SLE shaking in the form of a site-specific, uniform hazard acceleration response spectrum, with damping as indicated in Section 4.2.7. If nonlinear response history analysis is to be performed as part of the SLE evaluation, select ground motions and modify them to be compatible with the SLE spectrum in accordance with the recommendations of Chapter 3.

Commentary: SLE shaking is typically set at a return period of 43 years. Consequently, it can be reasonably expected that a tall building will be subjected to earthquake shaking at or exceeding this shaking level once or more during its service lifetime. This expectation has been taken into consideration in establishing the SLE criteria.

Considering that intended response of the building to SLE shaking is essentially elastic, the

Service Level EQ – Load Combinations

5.5.1 Load Combinations: Linear Modal Response Spectrum Analysis

Evaluate the structure for strength and drift for the following load combinations:

$$1.0D + 0.5L + 1.0E_x + 0.3E_y \quad (5-1)$$

$$1.0D + 0.5L + 0.3E_x + 1.0E_y \quad (5-2)$$

Take live load, *L*, as 80% of unreduced live loads that exceed 100 pounds per square foot (4.79 kN/m²) and 40% of other unreduced live loads.

Seismic Serviceability – Story Drift Limits

5.6.1 Story Drift Limit

Calculated story drift shall not exceed 0.5% of story height in any story, computed at extreme points for each floor plan in each of two orthogonal plan directions.

Commentary: The story drift limit of 0.5% for SLE shaking is intended to provide some protection of nonstructural components and also to ensure that permanent lateral displacement of the structure will be negligible. Extreme points on the floor plans are evaluated to help ensure that torsional response is properly controlled. It is important to understand that at a story drift of 0.5%, nonstructural damage, particularly for elements such as interior partitions, may not be negligible and considerable cosmetic repair may be required. Evaluation of damage to nonstructural elements can be performed using tools such as FEMA P58.

Deformation Controlled Actions

5.7 COMPONENT ACCEPTANCE CRITERIA - LINEAR ANALYSIS

5.7.1 Deformation-Controlled Actions

When response spectrum or linear response history analysis is used for the SLE evaluation, calculated demand-to-capacity ratios for deformation-controlled actions shall not exceed 1.5, where demand is calculated from load combinations in accordance with Section 5.5, and capacity is calculated as follows:

1. For reinforced concrete elements, the capacity is defined as the nominal strength in accordance with ACI 318 without applying the corresponding strength reduction factor ϕ .
2. For structural steel and composite steel and concrete elements, the capacity is defined as the nominal LRFD strength in accordance with AISC 341 and AISC 360, which is taken as the nominal strength without applying the corresponding resistance factor ϕ .

anticipated that expected strengths will be higher than the nominal strengths. Consequently, the demand-to-capacity ratio of 1.5 based on design strengths can be expected to result in only minor inelastic response.

These Guidelines do not provide more restrictive requirements for Risk Category III buildings in the SLE evaluation since the building code focuses primarily on limiting the probability of collapse for that Risk Category. If a higher level of certainty of meeting the service-level performance goals is desired for Risk Category III and IV buildings, a lower demand-to-capacity ratio for the deformation-controlled actions (for example, $1.5/1.25 = 1.2$ for Risk Category III buildings) can be applied.

Force Controlled Actions

5.7.2 Force-Controlled Actions

Calculated demand-to-capacity ratios for force-controlled actions shall not exceed 1.0, where demand is calculated from load combinations in accordance with Section 5.5, and capacity is calculated as follows:

1. For reinforced concrete elements and their connections, the capacity is defined as the design strength, taken as the nominal strength multiplied by the corresponding strength reduction factor ϕ in accordance with ACI 318.
2. For structural steel elements, for composite steel and concrete elements, and for their connections, the capacity is defined as the LRFD strength, which is taken as the nominal strength multiplied by the corresponding resistance factor ϕ in accordance with AISC 341 and AISC 360.

Commentary: *For force-controlled actions, strength reduction (resistance) factors ϕ of ACI 318, AISC 341, and AISC 360 and a demand-to-capacity ratio of 1.0 have been defined to promote a strength hierarchy in which yielding occurs in deformation-controlled actions before force-controlled actions. However, these factors alone may be insufficient to ensure that an appropriate yielding mechanism occurs. It is not uncommon, therefore, to design a*

Modeling Member Properties – Structural Steel

4.6.3 Effective Member Stiffness

Steel members and components: Model the elastic (initial) stiffness of steel members and components using full cross-sectional properties and the elastic modulus of steel [$E_s = 29,000$ ksi (200,000 MPa)].

Reinforced concrete components: In lieu of detailed justification, use the values in Table 4-3 for the effective stiffness of reinforced concrete members and components, along with component-specific guidance in subsections of Section 4.6.5.

Modeling Member Properties – Reinforced Concrete

Table 4-3 Reinforced concrete effective stiffness values.

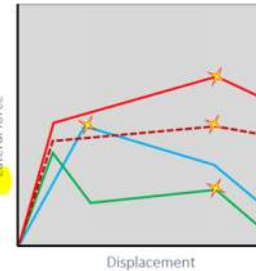
Component	Service-Level Linear Models			MCE _R -Level Nonlinear Models		
	Axial	Flexural	Shear	Axial	Flexural	Shear
Structural walls ¹ (in-plane)	$1.0E_cA_g$	$0.75E_cI_p$	$0.4E_cA_g$	$1.0E_cA_g$	$0.35E_cI_p$	$0.2E_cA_g$
Structural walls (out-of-plane)	--	$0.25E_cI_p$	--	--	$0.25E_cI_p$	--
Basement walls (in-plane)	$1.0E_cA_g$	$1.0E_cI_p$	$0.4E_cA_g$	$1.0E_cA_g$	$0.8E_cI_p$	$0.2E_cA_g$
Basement walls (out-of-plane)	--	$0.25E_cI_p$	--	--	$0.25E_cI_p$	--
Coupling beams with conventional or diagonal reinforcement	$1.0E_cA_g$	$0.07\left(\frac{L}{h}\right)E_cI_p$ $\leq 0.3E_cI_p$	$0.4E_cA_g$	$1.0E_cA_g$	$0.07\left(\frac{L}{h}\right)E_cI_p$ $\leq 0.3E_cI_p$	$0.4E_cA_g$
Composite steel / reinforced concrete coupling beams	$1.0(EA)_{trans}$	$0.07\left(\frac{L}{h}\right)EI_{trans}$	$1.0E_cA_{sw}$	$1.0(EA)_{trans}$	$0.07\left(\frac{L}{h}\right)EI_{trans}$	$1.0E_cA_{sw}$
Non-PT transfer diaphragms (in-plane only) ²	$0.5E_cA_g$	$0.5E_cI_p$	$0.4E_cA_g$	$0.25E_cA_g$	$0.25E_cI_p$	$0.1E_cA_g$
PT transfer diaphragms (in-plane only) ³	$0.8E_cA_g$	$0.8E_cI_p$	$0.4E_cA_g$	$0.5E_cA_g$	$0.5E_cI_p$	$0.2E_cA_g$
Beams	$1.0E_cA_g$	$0.5E_cI_p$	$0.4E_cA_g$	$1.0E_cA_g$	$0.3E_cI_p$	$0.4E_cA_g$
Columns	$1.0E_cA_g$	$0.7E_cI_p$	$0.4E_cA_g$	$1.0E_cA_g$	$0.7E_cI_p$	$0.4E_cA_g$
Mat (in-plane)	$0.8E_cA_g$	$0.8E_cI_p$	$0.8E_cA_g$	$0.5E_cA_g$	$0.5E_cI_p$	$0.5E_cA_g$
Mat ⁴ (out-of-plane)	--	$0.8E_cI_p$	--	--	$0.5E_cI_p$	--

P-Delta Analysis – A *Caution* for Seismic Design

4.2.8 P-Delta Effects

Include P-Delta effects in nonlinear analysis, regardless of whether elastic analysis design checks indicate that such effects are important. The P-Delta effects should include the destabilizing gravity loads for the entire building, where the gravity loads are spatially distributed to capture both building translation and twist.

Commentary: The widely used elastic stability coefficient ($\theta = P\delta/Vh$) is often an insufficient indicator of the importance of P-Delta effects in the inelastic range. P-Delta effects may become an overriding consideration when strength deterioration sets in and the tangent stiffness of the story shear force versus story drift relationship approaches zero or becomes negative. When this happens, the story drift ratchets, that is, it increases in one direction without the benefit of a full reversal that otherwise would straighten out the story. For this reason, and many others, realistic modeling of component deterioration and post-yield stiffness are critical aspects of modeling. The potential for dynamic instability is relatively high in flexible moment frame structures and in braced frames and shear wall structures in which one or several of the lower stories deform in a shear mode and the tributary gravity loads are large such that P-Delta will lead to a significant amplification of story drift demands. ATC 72 (2010) Chapter 2 provides detailed information on P-Delta effects and why and when the effects become an important consideration in inelastic structural response.



Positive slope can prevent instability

Negative slope can lead to instability

Summary

Design for stability is necessary

- Must address “Big Five” considerations, including:
 - 2nd-order effects
 - B_2 amplifier
 - » Or B_2 -based force (FOM)
 - Explicit second-order analysis
 - Stiffness-reduction effects
 - ELM: K
 - DM: Reduced stiffness ($0.8\tau_b$)
 - FOM: 0.8-based force; axial-force limit

Summary

Second-order effects are a key part of design for stability
 Second-order effects should be included in *all drift* evaluations
 Second-order effects can be included in the analysis or addressed via amplifiers (B_1 , B_2 and the new B_3)
 Inclusion of full system gravity load is necessary to capture $P\Delta$ effects

- Select appropriate load combination for reduced stiffness

Proper modeling is necessary to capture $P\delta$ effects

- Mesh members (4 segments) or use the B_1 amplifier

Design Amplifiers – You know about two of them

- The AISC B_1 Amplifier

$$B_1 = \frac{C_m}{1 - \alpha P_r / P_{e1}} \geq 1 \quad (\text{A-8-3})$$

- The AISC B_2 Amplifier

$$B_2 = \frac{1}{1 - \frac{\alpha P_{story}}{P_{e story}}} \geq 1 \quad (\text{A-8-6})$$

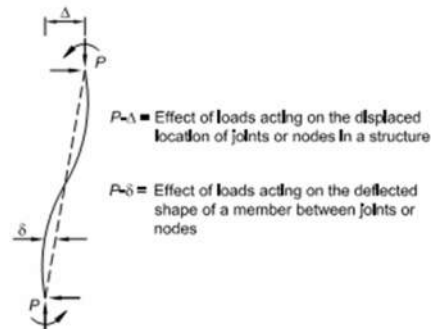


Fig. C-C2.1. $P-\Delta$ and $P-\delta$ effects in beam-columns.

And now, there is a third one – *THE B₃ AMPLIFIER*

The **B₃**
“SABELLI AMPLIFIER”

$$B_3 = \frac{0.8\tau_b}{1 - [1 - 0.8\tau_b]B_2}$$

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The **B₃**
“SABELLI AMPLIFIER”

$$B_3 = \frac{0.8\tau_b}{1 - [1 - 0.8\tau_b]B_2}$$

It's TASTY!

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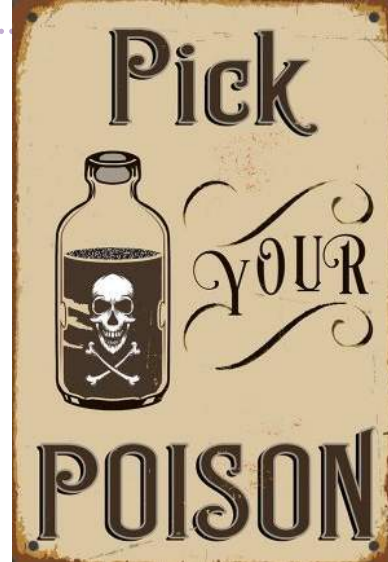
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Stability-design methods

	Braced Frames	Moment Frames
FOM	Additional lateral force Conservative	Additional lateral force Conservative
ELM		K factors Adjust for leaning columns
DM	Reduced-stiffness model	Reduced-stiffness model



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Indirect analysis method

Based on Direct analysis method

Full-stiffness model

$K=1$

There has to be a catch, right?

- Uses B_3 factor instead of reduced stiffness



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Equilibrium in the deformed condition: amplifier method

External lateral load H

Notional lateral load N

Total load effect on lateral system

$(H+N)L + \Sigma P\Delta_2$

Internal $P\Delta$ load effect

Total load effect on lateral system

$\bar{B}_2(H+N)$

Equilibrium @ $(\Delta=0) \rightarrow$ Equilibrium @ (Δ_2)

$H + \bar{H}_{P\Delta} = \bar{B}_2(H+N)$

First-order and second-order effects
(including stiffness reduction)

Amplified first-order analysis
(including imperfection loads)

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Equilibrium in the deformed condition : amplifier method

All deformations

Second-order effects

- $P\Delta$ effects
- $P\delta$ effects (system)

Geometric imperfections

Stiffness reduction due to inelasticity

Uncertainty in strength and stiffness

Δ_0

$\bar{B}_2(H+N)$

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Unpacking the amplifier

$\bar{B}_2(H + N)$

H First-order effects

N Initial imperfections

$\bar{B}_2 = B_2 B_3$


$B_2 = \frac{1}{1 - \frac{P_{story} \Delta_H}{R_M H L}}$

Second-order effects
 $P\delta$ and $P\Delta$

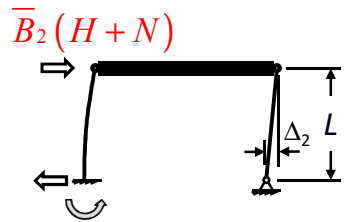
$B_3 = \frac{0.8\tau_b}{1 - [1 - 0.8\tau_b]B_2}$

Stiffness-reduction effects

$B_3 = \frac{4}{5 - B_2}$



$\bar{B}_2(H + N)$



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Indirect analysis method

Parameter τ_b

- Based on $\alpha P_r / P_{ns}$ ($= P_u / P_y$)
- Calculated member by member

Factor B_3

- Calculated story by story
 - Stiffness reduction of one member affects all members
- Applied story by story
 - Largest value can be applied to entire structure

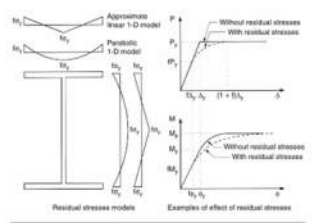


Figure 3.4 Schematic representation of self-equilibrating residual stresses in wide-flange structural shapes.

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2 ways to implement

	Pre-design	Loads	Analysis/ Design	Check	Iteration
Amplified 1st order analysis = 2nd order Analysis	Estimate B_2 and B_3 based on drift limit	Apply B_2 and B_3 to lateral forces	1 st order analysis	$\alpha P_r / P_{ns} \leq 0.5$	If required
Explicit 2nd order Analysis	Estimate B_2 and B_3 based on drift limit	Apply B_3 to lateral forces	2 nd order analysis	$\alpha P_r / P_{ns} \leq 0.5$	If required

Indirect analysis method

Two ways to implement

- **Amplified 1st-order analysis**
 - Second-order effects approximated by B_2
 - Stiffness reduction using B_3 based on B_2
 - Both can be determined based on drift limit in advance of design

$$\bar{B}_2(H + N) = B_2 B_3 (H + N)$$

$\alpha P_r / P_{ns}$	≤ 0.5	0.55	0.6	0.65	0.7
τ_b	1	0.99	0.96	0.91	0.84
B_2	$B_2 B_3$				$B_2 t$
1.00	1.00	1.00	1.00	1.00	1.00
1.05	1.06	1.06	1.07	1.07	1.08
1.10	1.13	1.13	1.13	1.14	1.16
1.15	1.19	1.20	1.20	1.22	1.24
1.20	1.26	1.27	1.28	1.30	1.33
1.25	1.33	1.34	1.35	1.38	1.42
1.30	1.41	1.41	1.43	1.46	1.52
1.35	1.48	1.49	1.51	1.55	1.63
1.40	1.56	1.56	1.59	1.65	1.74
1.45	1.63	1.64	1.68	1.74	1.86
1.50	1.71	1.73	1.77	1.84	1.98

Indirect analysis method

Two ways to implement

- **Explicit 2nd-order analysis**
 - 2nd-order effects using software
 - Stiffness reduction using B_3
 - based on approximate B_2
 - » based on drift limit
 - in advance of design
 - Or Δ_2/Δ_1 (or H_2/H_1) from analysis

$$\bar{B}_2(H + N) = SOA[B_3(H + N)]$$

$\alpha P_r/P_{ns}$	≤ 0.5	0.55	0.6	0.65	0.7
τ_b	1	0.99	0.96	0.91	0.84
B_2			B_3		B
1.00	1.00	1.00	1.00	1.00	1.00
1.05	1.01	1.01	1.02	1.02	1.03
1.10	1.03	1.03	1.03	1.04	1.05
1.15	1.04	1.04	1.05	1.06	1.08
1.20	1.05	1.06	1.06	1.08	1.11
1.25	1.07	1.07	1.08	1.10	1.14
1.30	1.08	1.09	1.10	1.13	1.17
1.35	1.10	1.10	1.12	1.15	1.21
1.40	1.11	1.12	1.14	1.18	1.24
1.45	1.13	1.13	1.16	1.20	1.28
1.50	1.14	1.15	1.18	1.23	1.32

Advantages and disadvantages: IAM and FOM



	Advantages of IAM	Disadvantages of IAM
FOM	<ul style="list-style-type: none"> • Lower forces • $\alpha P_r/P_{ns} \leq 0.7$ for moment-frame columns • $B_2 \leq 2.0$ • No additional lateral loads • $B_2 B_3$ indicates stability 	<ul style="list-style-type: none"> • Second-order analysis <p><i>(But FOM is limited to $B_2 \leq 1.5$....)</i></p>

Advantages and disadvantages: IAM and ELM



	Advantages of IAM	Disadvantages of IAM
ELM	<ul style="list-style-type: none"> • Lower demand-to-capacity ratios (moment frames) • $K=1$ • $B_2 \leq 2.0$ • Provides appropriate design forces for connections & beams • B_3 indicates stability 	<ul style="list-style-type: none"> • B_3 amplifier • $\alpha P_r / P_{ns} \leq 0.7$ for moment-frame columns

Advantages and disadvantages: IAM and DM



	Advantages of IAM	Disadvantages of IAM
DM	<ul style="list-style-type: none"> • Single calculation of B_3 • One model • B_3 is an indicator of stability 	<ul style="list-style-type: none"> • Potentially higher forces (largest τ_b) • Limited to <ul style="list-style-type: none"> ○ Vertical columns ○ $B_2 \leq 2.0$ ○ $\alpha P_r / P_{ns} \leq 0.7$ for moment-frame columns



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Second-order analysis
 Let's set aside N & B_3 and focus on B_2

$$HL + \Sigma P \Delta_2 = B_2 HL$$

$$B_2 = \frac{HL + P_{story} \Delta_2}{HL} = 1 + \frac{P_{story} \Delta_2}{HL}$$

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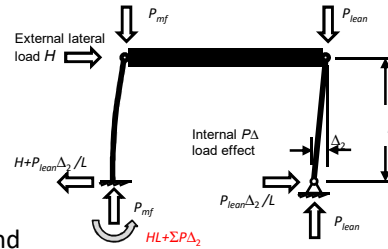
Second-order analysis

$$B_2 = 1 + \frac{P_{story} \Delta_2}{HL}$$

Equilibrium at Δ_2

$$B_2 = \frac{1}{1 - \frac{P_{story} \Delta_1}{R_M HL}}$$

Force amplifier with $P\Delta$ and R_M stiffness reduction



$$1 + \frac{P_{story} \Delta_2}{HL} = \frac{1}{1 - \frac{P_{story} \Delta_1}{R_M HL}}$$

$$\Delta_2 = \frac{B_2}{R_M} \Delta_1$$

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Second-order analysis

$$1 + \frac{P_{story} \Delta_2}{HL} = \frac{1}{1 - \frac{P_{story} \Delta_1}{R_M HL}}$$

$$\frac{H}{\Delta_2} = R_M \frac{H}{\Delta_1} - \frac{P_{story}}{L}$$

- Effective stiffness
 - 2nd order

$$\frac{H}{\Delta_1} = \frac{1}{R_M} \left(\frac{H}{\Delta_2} + \frac{P_{story}}{L} \right)$$

- Mechanical stiffness
 - 1st order

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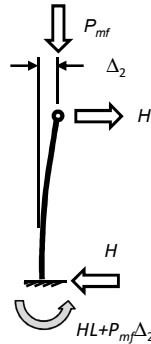
PΔ stiffness reduction

$$B_2 = \frac{1}{1 - \frac{HL/\Delta_1 - (12/\pi^2 - 1)P_{mf}}{P_{story}}}$$

LeMessurier, 1977

$$R_M = 1 - \frac{P_{mf}\Delta_1}{HL} \left(\frac{12}{\pi^2} - 1 \right)$$

$$R_M = 1 - 0.216\theta \frac{P_{mf}}{P_{story}}$$



$$\leq \frac{1}{1 - \frac{P_{story}\Delta_1}{\left(1 - 0.15 \frac{P_{mf}}{P_{story}}\right)HL}}$$

AISC Specification

$$R_M = 1 - 0.15 \frac{P_{mf}}{P_{story}}$$

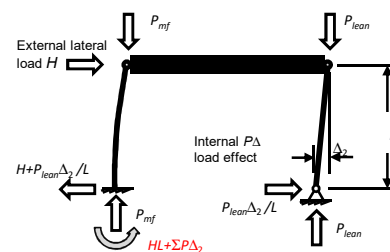
AISC parameter R_M
Calibrated to $\theta = 0.7$

Second-order analysis

$$B_2 = 1 + \frac{P_{story}\Delta_2}{HL} \quad B_2 = \frac{1}{1 - \frac{P_{story}\Delta_1}{R_M HL}}$$

$$1 + \frac{P_{story}\Delta_2}{HL} = \frac{1}{1 - \frac{P_{story}\Delta_1}{R_M HL}}$$

$$\Delta_2 = \frac{B_2}{R_M} \Delta_1 \quad R_M = 1 - \frac{P_{mf}\Delta_1}{HL} \left(\frac{12}{\pi^2} - 1 \right)$$



Based on LeMessurier, 1977

B_2 with AISC $R_M \sim$ true B_2/R_M for:
 $\theta \leq 0.3$ with $P_{mf}/P_{story} = 1.0$
 $\theta \leq 0.5$ with $P_{mf}/P_{story} = 1/3$

$$R_M = 1 - 0.15 \frac{P_{mf}}{P_{story}}$$

Second-order analysis

$$\frac{H}{\Delta_2} = R_M \frac{H}{\Delta_1} - \frac{P_{story}}{L} = \frac{H}{\Delta_1} - \frac{P_{story} + \left(\frac{12}{\pi^2} - 1\right) P_{mf}}{L}$$

$$\frac{H}{\Delta_1} = \frac{1}{R_M} \left(\frac{H}{\Delta_2} + \frac{P_{story}}{L} \right) = \frac{H}{\Delta_2} + \frac{P_{story} + \left(\frac{12}{\pi^2} - 1\right) P_{mf}}{L}$$

$$\frac{P_{story} + \left(\frac{12}{\pi^2} - 1\right) P_{mf}}{L} = \frac{P_{lean} + \frac{12}{\pi^2} P_{mf}}{L}$$

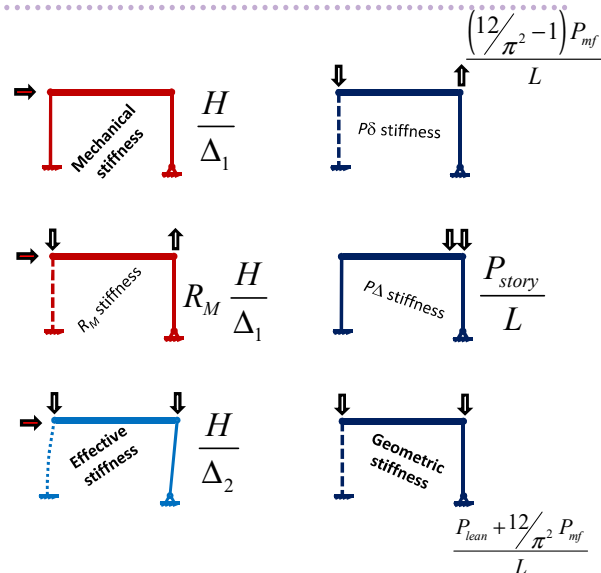
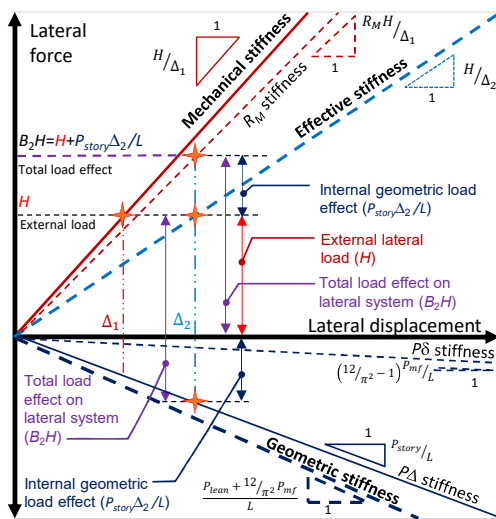
- Effective stiffness
 - 2nd order
- Mechanical stiffness
 - 1st order
- Geometric stiffness
 - $P\Delta$ and $P\delta$

$P\Delta$ effect		$P\delta$ effect
$P_{lean}\Delta_2/L$	$P_{mf}\Delta_2/L$	$(12\pi^2-1)P_{mf}\Delta_2/L$
Leaning columns		Moment-frame columns

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Second-order analysis



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Determine B_2 in advance of design

$$B_2 = 1 + \frac{P_{story} \Delta_2}{HL}$$

$$\Delta_2 = \frac{\Delta_{all}}{C_d}$$

$$B_2 = 1 + \left[\frac{P_{story}}{C_d H} \right] \left[\frac{\Delta_{all}}{L} \right]$$

$C_d = 1.0$ for wind

H is load corresponding to Δ_{all}/C_d
(This is a stiffness term!)

$P_{story}/C_d H$		2	4	5	6.7	8	10	15	20	25	33	40	50	60	80	100
Δ_{all}/L	Δ_{all}	Values of B_2														
0.0025	L/400	1.01	1.01	1.01	1.02	1.02	1.03	1.04	1.05	1.06	1.08	1.10	1.13	1.15	1.20	1.25
0.0050	L/200	1.01	1.02	1.03	1.03	1.04	1.05	1.08	1.10	1.13	1.17	1.20	1.25	1.30	1.40	1.50
0.0100	L/100	1.02	1.04	1.05	1.07	1.08	1.10	1.15	1.20	1.25	1.33	1.40	1.50	1.60	1.80	2.00
0.0150	L/67	1.03	1.06	1.08	1.10	1.12	1.15	1.23	1.30	1.38	1.50	1.60	1.75	1.90	2.20	2.50
0.0200	L/50	1.04	1.08	1.10	1.13	1.16	1.20	1.30	1.40	1.50	1.67	1.80	2.00	2.20	2.60	3.00
0.0250	L/40	1.05	1.10	1.13	1.17	1.20	1.25	1.38	1.50	1.63	1.83	2.00	2.25	2.50	3.00	3.50

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Determine B_2 and B_3 in advance of design

$$B_2 = 1 + \frac{P_{story} \Delta_2}{HL}$$

$$B_2 = 1 + \frac{P_{story} \Delta_{all}}{C_d HL}$$

H is load corresponding to Δ_{all}/C_d

$$B_3 = \frac{0.8\tau_b}{1 - [1 - 0.8\tau_b] B_2}$$

$\frac{P_{story} \Delta_{all}}{C_d HL}$	B_2	$P_u/P_y \leq 0.5$				
		0.55	0.60	0.65	0.70	
0.025	1.03	1.01	1.01	1.01	1.01	
0.050	1.05	1.01	1.01	1.02	1.02	
0.075	1.08	1.02	1.02	1.02	1.03	
0.100	1.10	1.03	1.03	1.03	1.04	
0.125	1.13	1.03	1.03	1.04	1.05	
0.150	1.15	1.04	1.04	1.05	1.06	
0.175	1.18	1.05	1.05	1.06	1.07	
0.200	1.20	1.05	1.06	1.06	1.08	
0.225	1.23	1.06	1.06	1.07	1.09	
0.250	1.25	1.07	1.07	1.08	1.10	
0.275	1.28	1.07	1.08	1.09	1.11	
0.300	1.30	1.08	1.09	1.10	1.13	
0.325	1.33	1.09	1.09	1.11	1.14	

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Summary

Mechanical, effective, and geometric stiffness are related

- Geometric stiffness includes $P\Delta$ and $P\delta$ effects
- Effective stiffness is true stiffness in presence of vertical loads
- Effective = Mechanical – Geometric

Stability is based on equilibrium in the deformed condition

- Drift limits correspond to the deformed condition
- Stability amplifiers can be determined from the drift limit

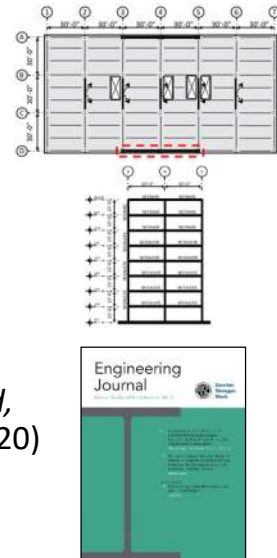
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Design example

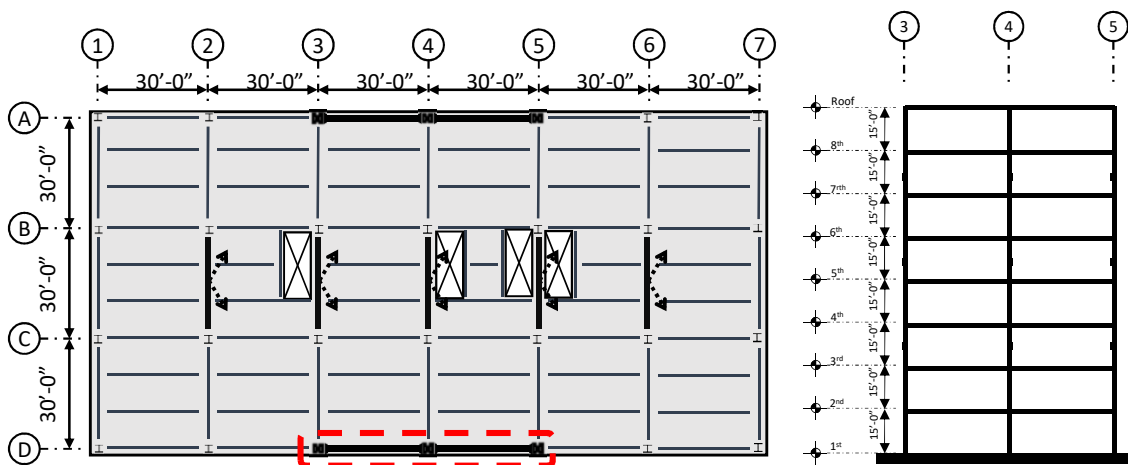
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Example design

- 8-story moment frame
- Indirect analysis method
 - Compare results with Direct Analysis
- Wind design (no seismic)
- Size members for drift
- Check strength/stability
- Factors determined based on drift limit
 - (Appendix B of *Indirect Analysis Method*, AISC Engineering Journal, Quarter 2, 2020)



Plan and elevation



Vertical loads

General					
Story Height		Dead Load	Live Load	$R_M = 1 - 0.15 \frac{P_{mf}}{P_{story}}$	
Level	L (in.)	Dead Load (kip)	Live Load (kip)	P_{mf}/P_{story}	R_M
8	180	2000	0	0.275	0.959
7	180	2000	1600	0.275	0.959
6	180	2000	1600	0.275	0.959
5	180	2000	1600	0.275	0.959
4	180	2000	1600	0.275	0.959
3	180	2000	1600	0.275	0.959
2	180	2000	1600	0.275	0.959
1	180	2000	1600	0.275	0.959

All pre-design information: **no preliminary member design**



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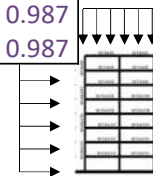
Drift design

Drift		$L/400$		$\frac{H}{\Delta_1}$		$R_M = 1 - \frac{P_{mf}\Delta_1}{HL} \left(\frac{12}{\pi^2} - 1 \right)$		
		$1.0DL+0.25LL$						
Level	$H_{service}$ (kip)	$\Delta_{allowable}$ (in.)	$\frac{H}{\Delta_2}$	P_{story} (kip)	$K_{required}$ (kip/in)	Δ_1 (in.)	B_2	R_M
8	20.0	0.450	Δ_2	2,000	56	0.357	1.26	0.988
7	40.0	0.450		4,400	114	0.350	1.29	0.987
6	60.0	0.450		6,800	173	0.347	1.30	0.987
5	80.0	0.450		9,200	231	0.346	1.30	0.987
4	100.0	0.450		11,600	289	0.345	1.30	0.987
3	120.0	0.450		14,000	348	0.345	1.30	0.987
2	140.0	0.450		16,400	406	0.345	1.31	0.987
1	160.0	0.450		18,800	464	0.344	1.31	0.987

All pre-design information: **no preliminary member design**

Technically correct Used in paper

$$\frac{H}{\Delta_1} \geq \frac{1}{R_M} \left(\frac{H}{\Delta_2} + \frac{\sum P}{L} \right) \approx \frac{H}{\Delta_2} + \frac{\sum P}{R_M L}$$



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Strength/Stability check

aP_r/P_{ns}	τ_b	Δ_1 (in.)	Δ_2 (in.)	B'_2	B'_3
0.07	1.00	0.39	0.49	1.25	1.07
0.16	1.00	0.59	0.79	1.35	1.10
0.15	1.00	0.54	0.74	1.36	1.10
0.21	1.00	0.57	0.79	1.38	1.11
0.22	1.00	0.56	0.78	1.39	1.11
0.28	1.00	0.60	0.84	1.41	1.11
0.25	1.00	0.56	0.78	1.38	1.11
0.29	1.00	0.45	0.59	1.32	1.09

Second-order analysis Correct stiffness
 $B'_2 = \Delta_2/\Delta_1$ Correct R_M via
 B'_3 calculated from B'_2 proper $P\delta$
 (for comparison only) modeling

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Strength/Stability check

Beam check			Column check		
IAM Beam (DCR)	DM Beam (DCR)	Ratio IAM/DM	IAM Column (DCR)	DM Column (DCR)	Ratio IAM/DM
0.13	0.12	1.06	0.14	0.14	1.04
0.21	0.20	1.04	0.45	0.45	1.01
0.31	0.30	1.02	0.25	0.25	1.02
0.27	0.27	1.02	0.45	0.45	1.01
0.32	0.31	1.02	0.47	0.47	1.01
0.29	0.29	1.01	0.58	0.58	1.01
0.33	0.32	1.01	0.50	0.50	1.01
0.33	0.32	1.03	0.65	0.64	1.02

IAM: Indirect analysis method
 DM: Direct analysis method

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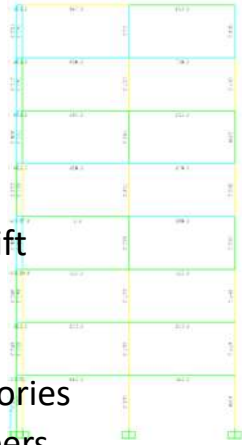
Example summary

Indirect analysis method is easy to implement

- One model
- $K=1$

Indirect analysis method is conservative

- Factors determined based on drift limit
 - Could be re-calculated based on actual drift
- Specification R_M is conservative
- Worst case B_3 applied all stories
 - Lower values could be applied at upper stories
- Conservatism does not result in larger members



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Summary

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Summary

Design for stability is necessary

- Must address “Big Five” considerations, including:
 - 2nd-order effects
 - B_2 amplifier
 - » Or B_2 -based force (FOM)
 - Explicit second-order analysis
 - Stiffness-reduction effects
 - ELM: K
 - DM: Reduced stiffness ($0.8\tau_b$)
 - FOM: 0.8-based force; axial-force limit

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Summary

Indirect Analysis Method: useful option for moment frames

- $K=1$
- One model
- B_3 addresses stiffness-reduction effects
 - $0.8\tau_b$
 - Function of B_2

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Summary

Second-order effects part of design for stability

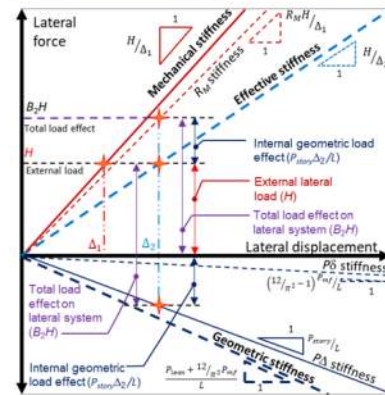
- included in the analysis
- or addressed via amplifiers

Mechanical, effective, and geometric stiffness

- Geometric stiffness includes $P\Delta$ and $P\delta$
- Effective stiffness is true stiffness
- Effective = Mechanical – Geometric

Equilibrium in the deformed condition

- Drift limits correspond to the deformed condition
- Amplifiers can be determined from drift limit



Thank you

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Questions?

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